



# **Observations, Ideas, and Opinions**

## **Systems Engineering and Integration for Return to Flight**

**George K. Gafka**

**Project Management Challenge Conference**

**March, 2006**





# Observations, Ideas, and Opinions

- **Upfront Disclaimer #1**

- Material transmitted in this presentation may not represent the opinion or policy of NASA!

- **Upfront Disclaimer #2**

- Presenter is conveying some very contextual examples of personal experiences which are not meant to be interpreted as the absolute truth or the right answer for everyone or every situation!

**Process/digest the material as you see fit and decide what may be worth taking away.**



# Observations, Ideas, and Opinions Presentation Outline

You are here.

- **Project Management & Systems Engineering Challenges**
  - In The Beginning...What is your mission? Can you “certify” to it?
  - Team Roles/Responsibilities/Requirements/Contracts/Deliverables
  - Use-As-Is becomes most critical capability!
    - Flight History Database, a surprisingly contentious topic
  - Tile Repair is really tough, becomes “best effort” for RTF
    - Killer/“Golden” Requirements: Bubbles
    - Tough Trade Spaces
  - Delivery for RTF
- **STS-114**
- **Conclusion**

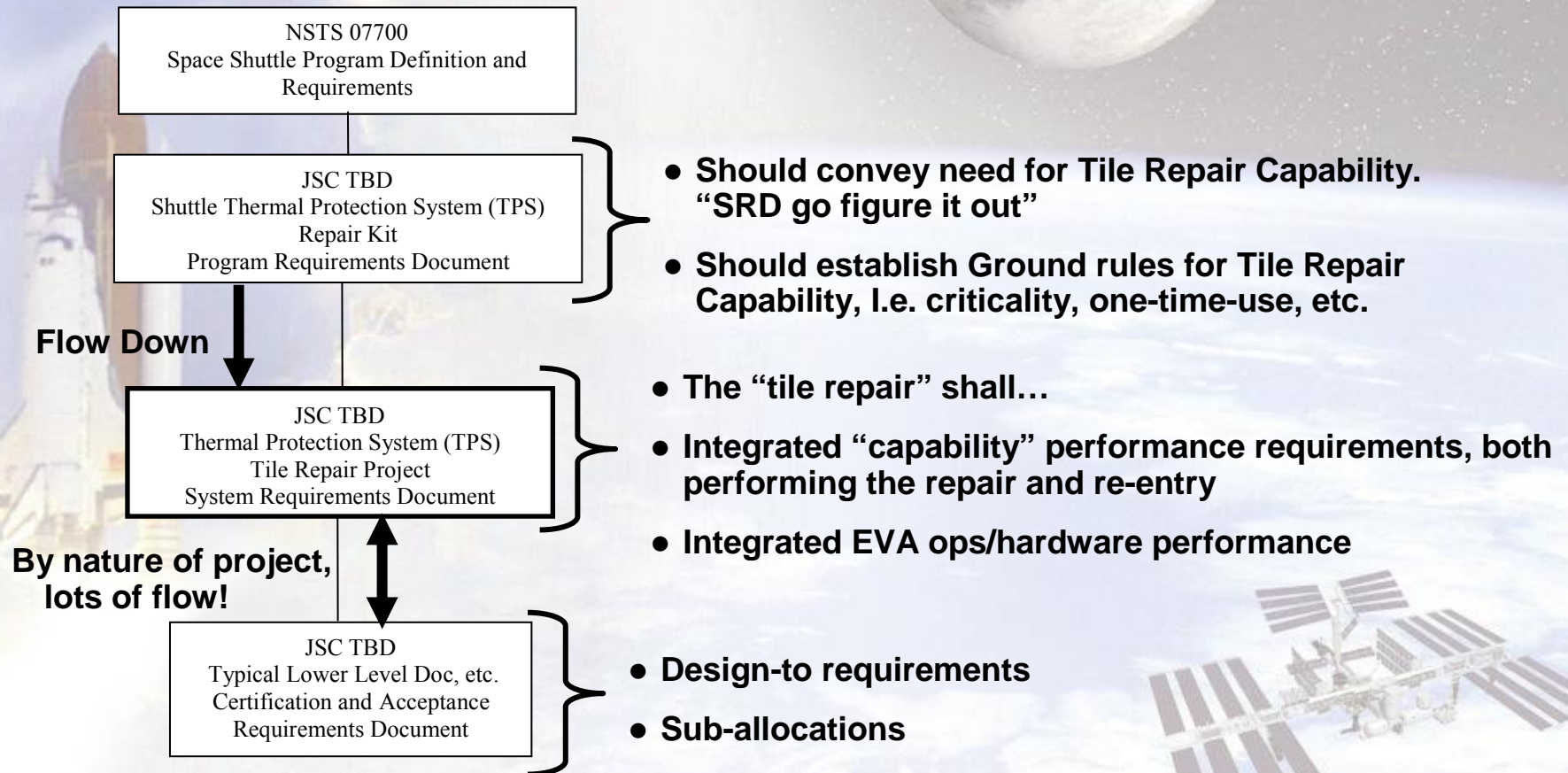
- **Understanding/Influencing/Accepting Your Environment**
  - Cost, schedule, technical/safety, political, emotional
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- **Effective People Skills and Communication, a key to success!**
  - Integrity/credibility
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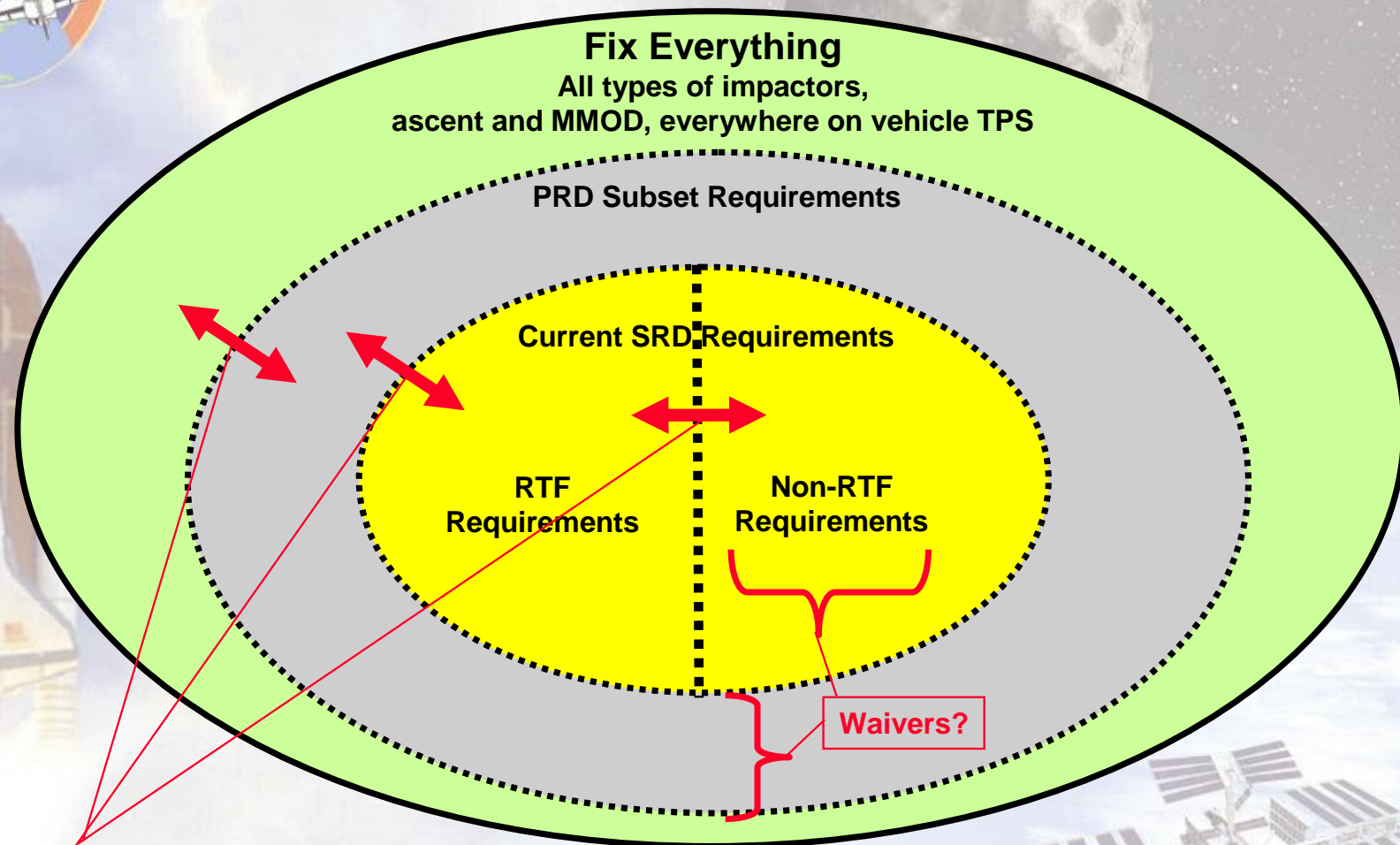
# In The Beginning... Project Documentation Philosophy

## Thermal Protection System (TPS) Tile Repair Project Documentation Tree





# REQUIRED Communication with Program Requirements Flow and Philosophy



Who is responsible to set boundary and accept risk?  
Who is responsible to substantiate boundary?  
MA, MS, MV?



# Tile Repair Project RTF Mission

- **Per our revised SRD and Verification Plan, the Tile Repair Project is responsible for delivering the capability to:**
  - Assess tile damage locations and provide near real-time technical rationale to support “Use-as-is” disposition
  - Provide repair materials (qualified vendor), physical tools and operational techniques to conduct a developmental DTO and constitute an emergency tile repair capability if needed
  - Document Limited material and system level test results
- **The Tile Repair Project is responsible for validating the PRD inspection requirements for size of tile damage not requiring inspection by OBSS**
  - 3” for acreage tile
  - 1” for tiles near door penetrations

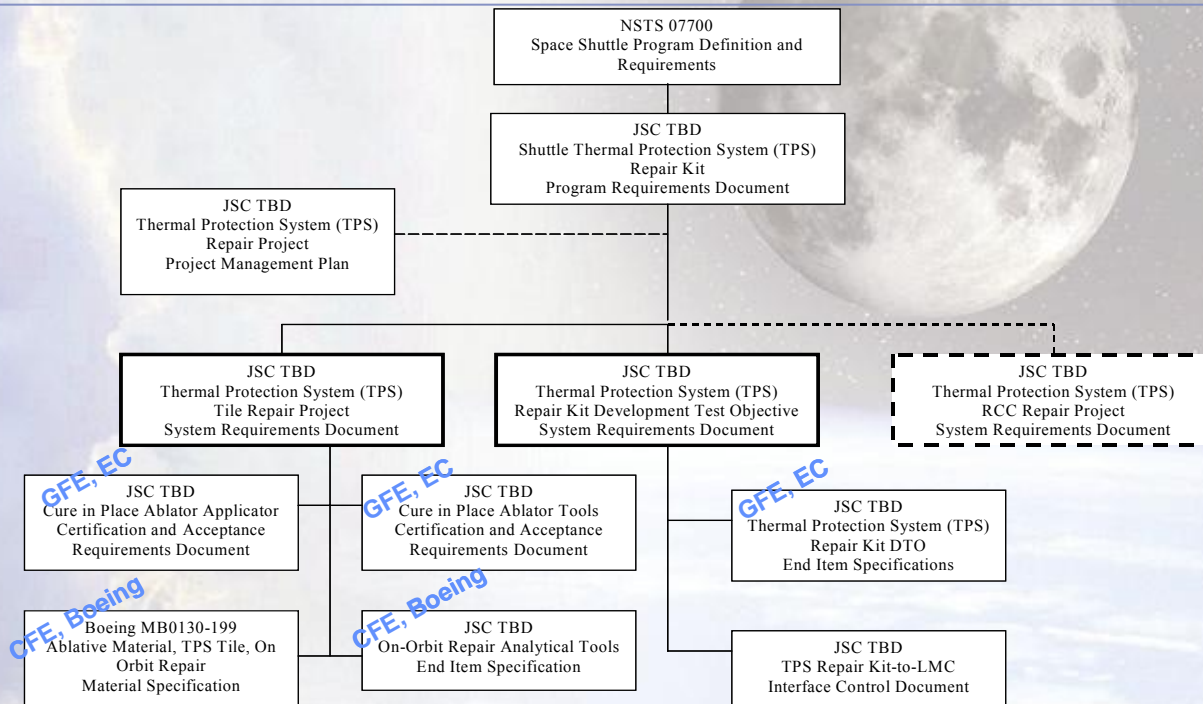
**NOTE: We should think of our “Use-As-Is” capability being comprised of two parts:  
Analytical Tools & Flight History Database!!!**





# Planned TRP Deliverables/Documentation

## Thermal Protection System (TPS) Tile Repair Project Documentation Tree



### CFE

- “Use-as-is” Analytical Tools (USA/Boe)
  - Cavity Heating Tool
  - Catalytic Heating Tool: Damaged
  - 3D Acreage Tile Thermal Tool
  - Special Config. Thermal Models
  - Tile Stress Tool – RTV Bondline (45 deg)
  - Stress Assessor Tool
- Repair Materials (USA/Boe/LM/OSS)
  - STA-54
  - EW

### GFE

- EVA Hardware (JSC EC/XA)
  - EVA Repair Mat'l Aplicators
  - EVA Handtools
- “Use-as-is” Analytical Tools
  - CFD for Cavity Heating: Baseline (Ames)
  - CFD for Cavity Heating: Flt Trace. (Ames)
  - Boundary Layer Transition Predict. (LaRC)



# TRP - Roles/Responsibilities Repair Material

**Process Dev.  
SE&I IPT**

**Prod/Logistics IPT  
(LM, USA/KSC)**

**USA  
P.O.**

Flt 3+ Prod.

**NASA  
IWTA (GFE)**

R&D  
Pre-qual testing  
Mat'l down-select  
Scale-up  
System level testing  
KC-135 & HTV testing

**NASA Project  
MV, EA, ES, EC**



**LMSSC -  
AO**

Material developer  
Material provider  
Material testing  
Tool provider (LMSO)  
Mat'l-Canister-Tool C/O

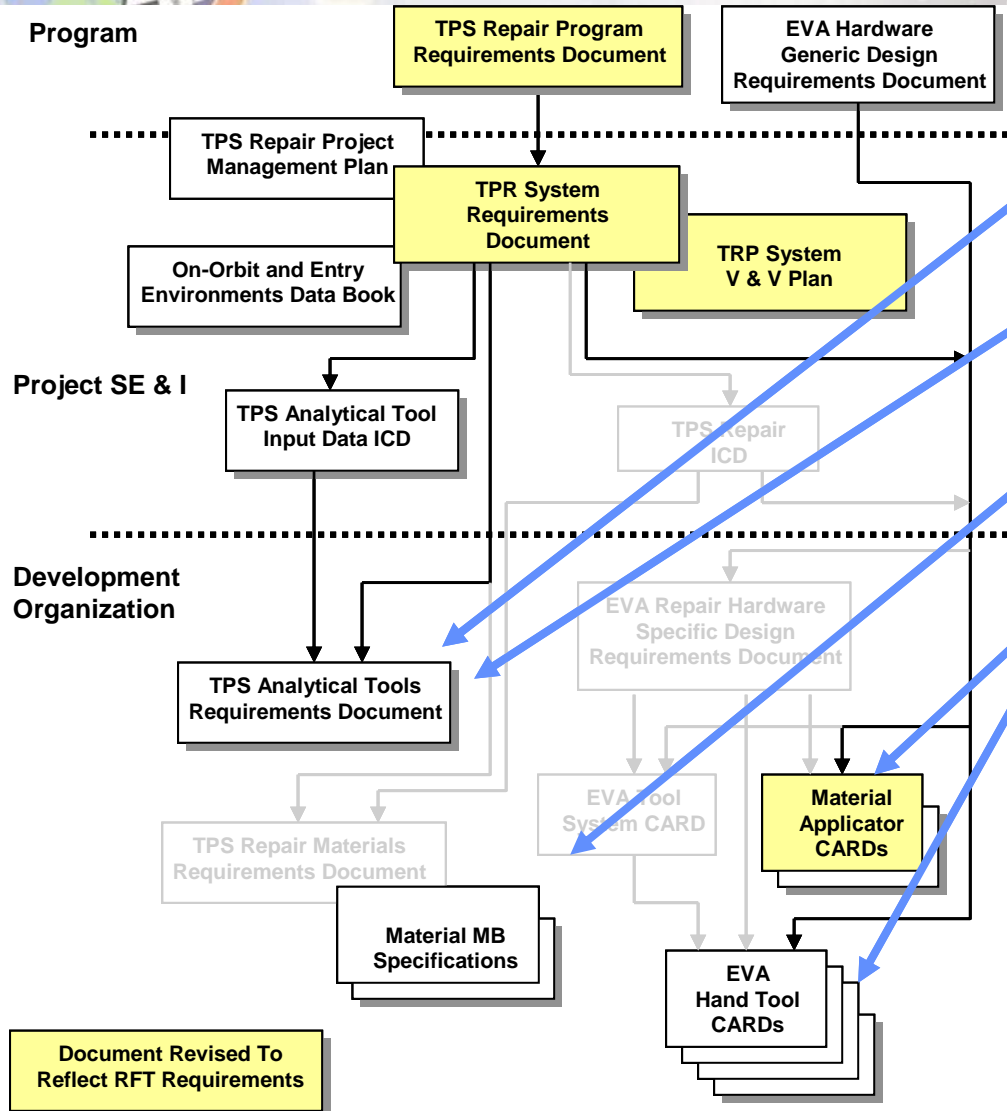
**Boeing  
P.O. (CFE)**

Characterization  
Qualification  
Verification  
HTV-2  
Flt 1 & 2 Production





# Planned TRP Documentation For RTF



- **Verified analytical tools for damaged acreage tile**
- **Validation of damage size inspection requirement**
- **Repair materials qualified to Material Specifications (physical properties and processes)**
- **EVA tools verified for Crit 3 safety**
- **Limited material and system level test data**



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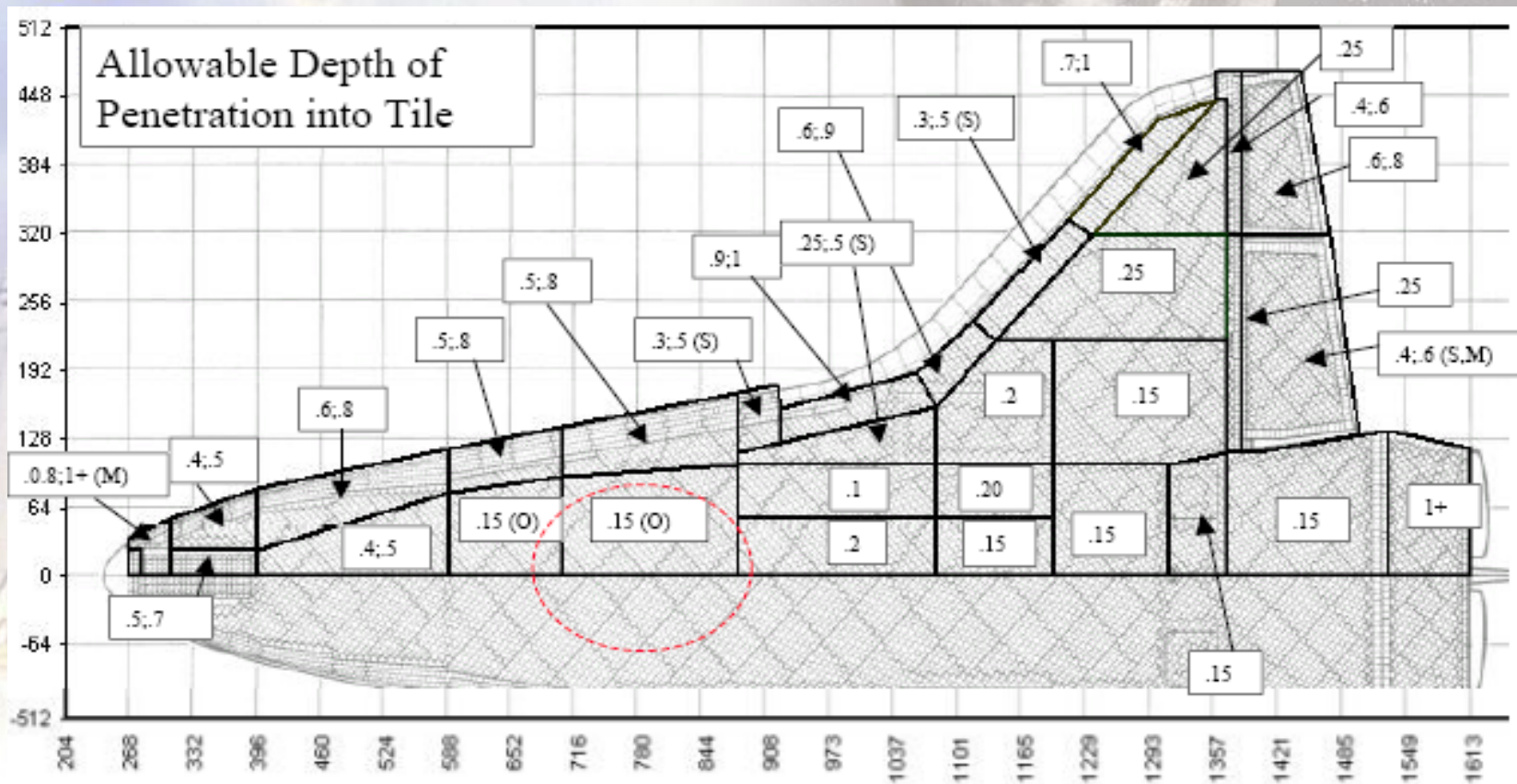
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# Results to date (early 2005) Best Estimate of Damage “Map”







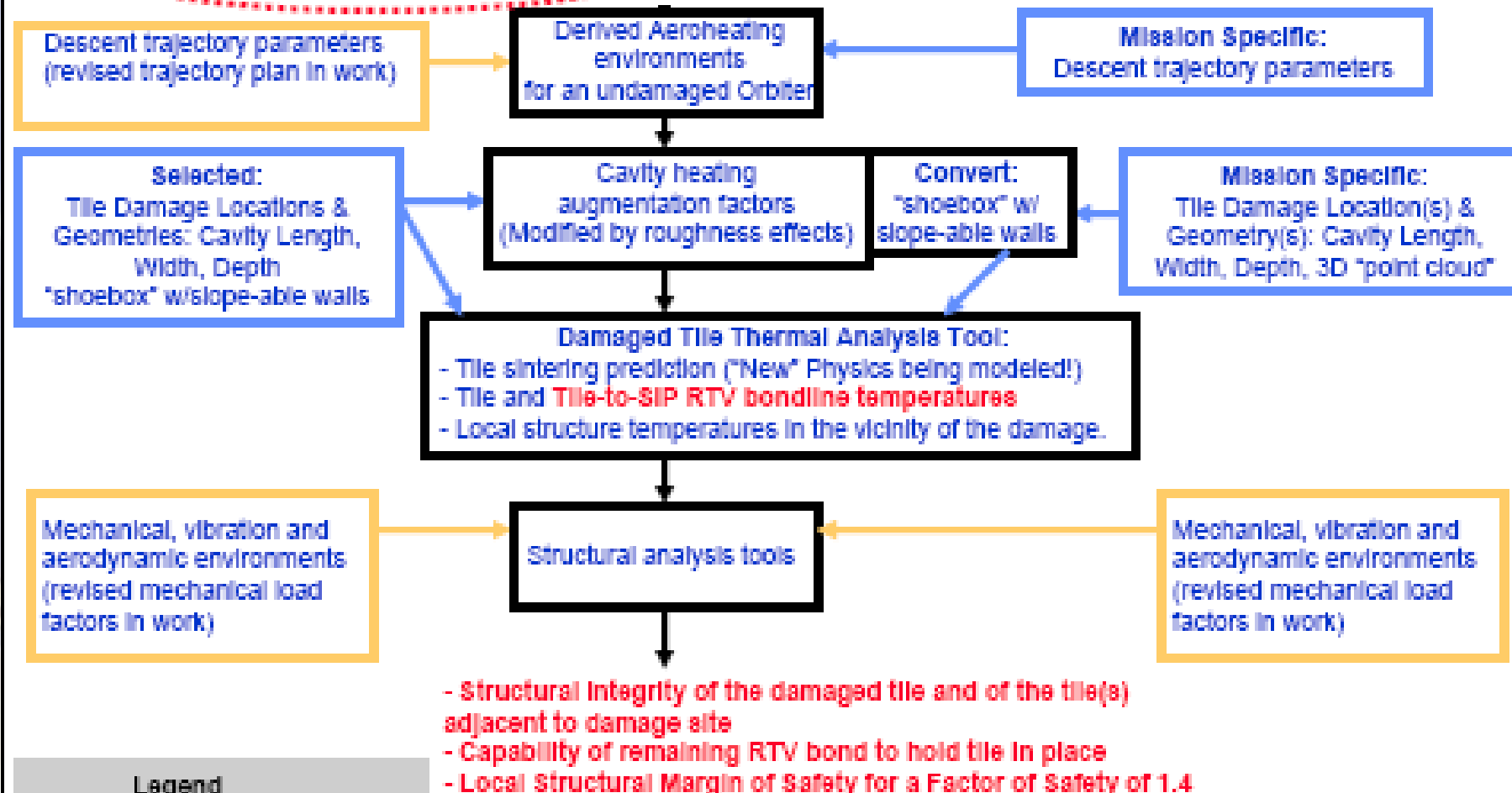
# Development of TRP "Use-As-Is" Analytical Tools

## Development of Inspection Criteria (need for OBSS/depth)



### Pre-Flight Development

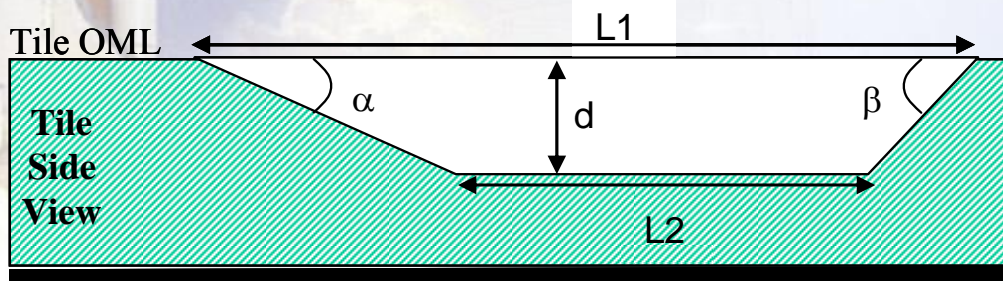
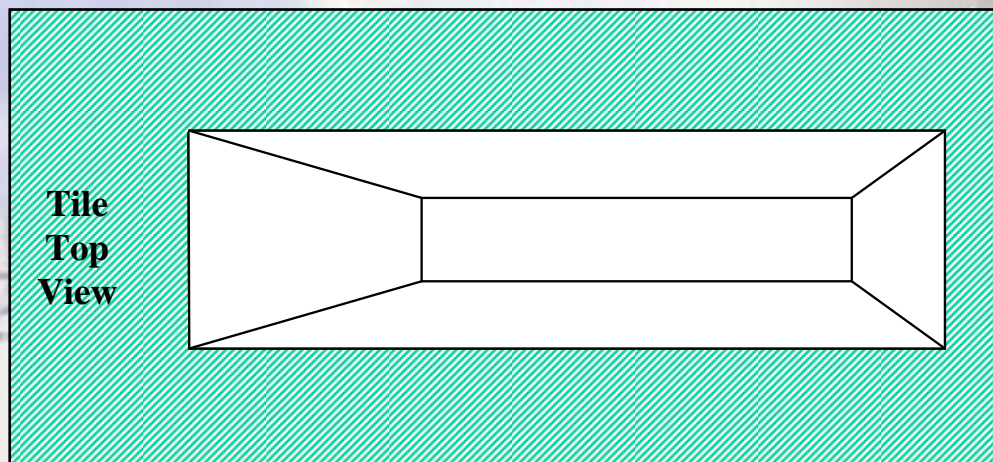
### Real-Time Mission Specific Use





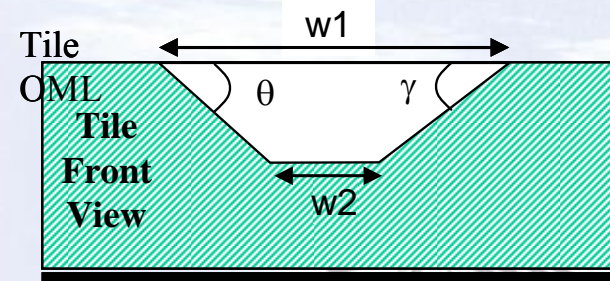
# In-Scope Damage Geometries

**Flow Direction**



Underlying Orbiter Structure

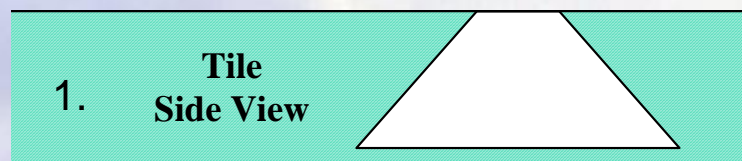
**Limitations:**  
 $TBD \leq L1 \leq 20$   
 $TBD \leq L2 \leq 20$   
 $(L1 \geq L2)$   
 $0.02 \leq d \leq \text{full tile}$   
 $0.25 \leq w1 \leq 10$   
 $0.25 \leq w2 \leq 10$   
 $(w1 \geq w2)$   
 $0 \leq \alpha \leq 90$   
 $0 \leq \beta \leq 90$   
 $0 \leq \theta \leq 90$   
 $0 \leq \gamma \leq 90$



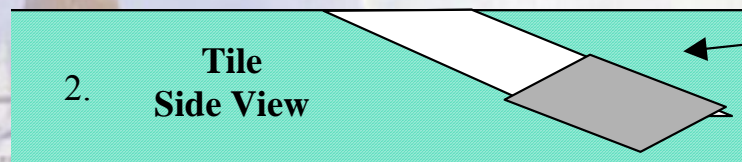
Underlying Orbiter Structure



# Examples of Out-of-Scope Damage Types/Geometries



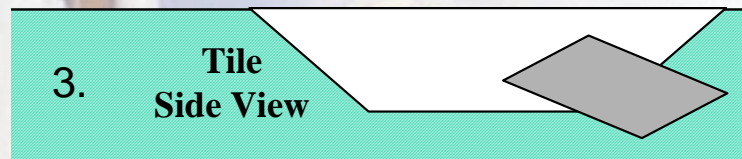
Damage geometry out-of-scope ( $w2 > w1$ , represents damage from certain MMOD impacts)



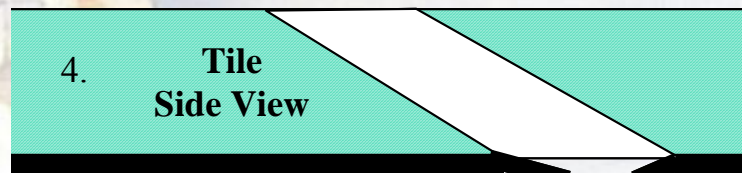
Damage geometry out-of-scope ( $\beta$  constraint violation, represents damage from certain high density impactors, i.e. ablator material)

**And/or**

Impactor remaining in cavity



Impactor remaining in cavity



Underlying Orbiter Structure

Penetration into structure  
(with possible underlying structural damage)

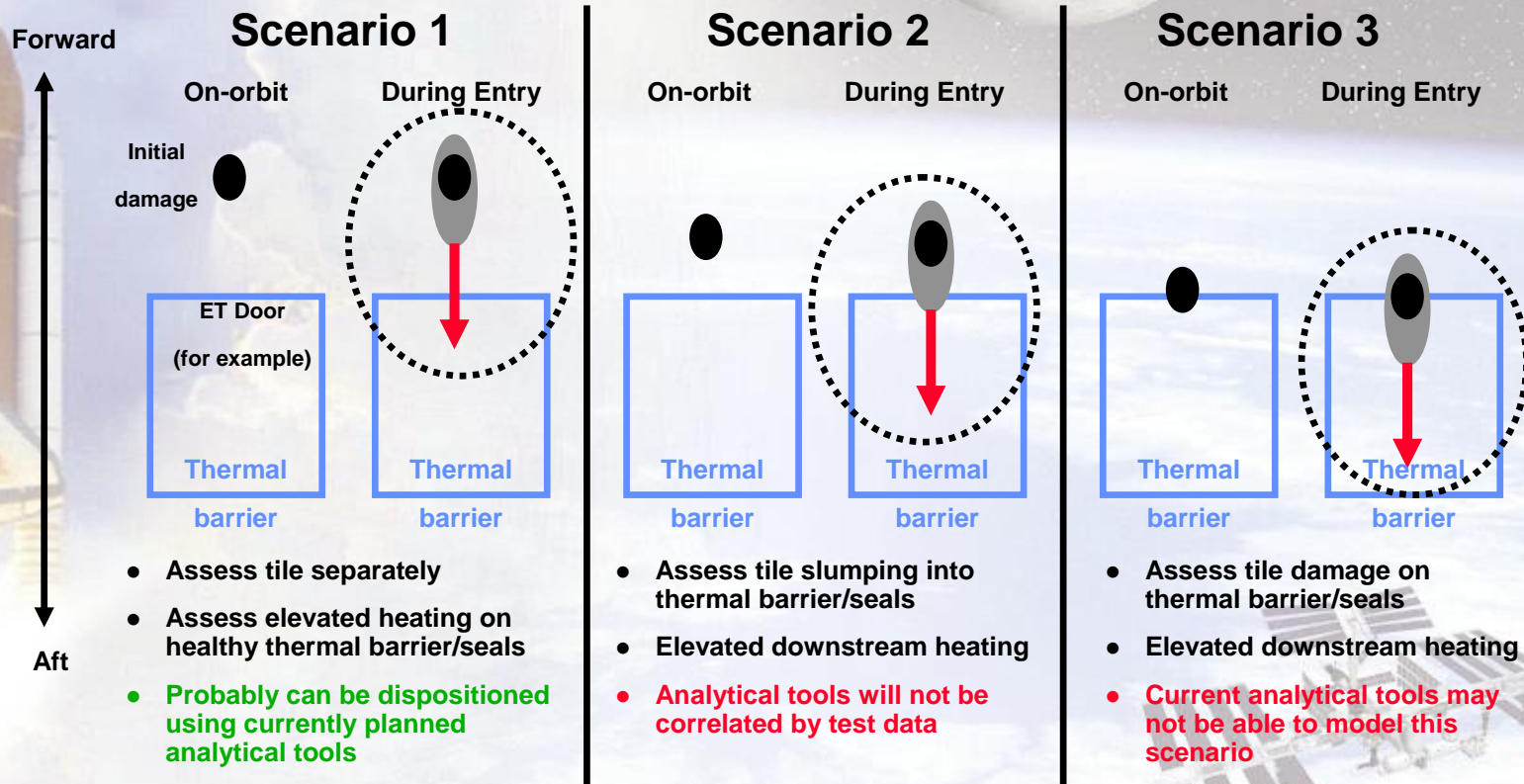




# NLGD,MLGD,ETD

## Notional Depiction of Capability/Concern

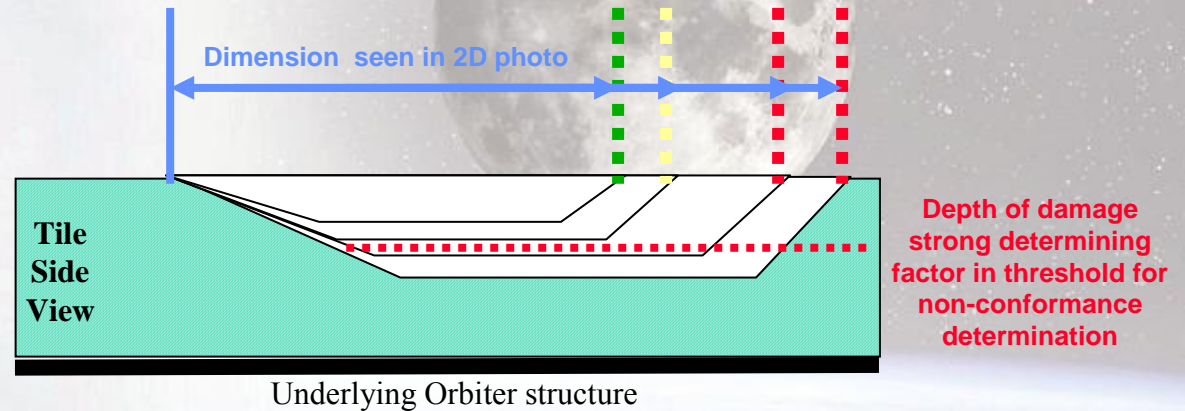
- NOTE: Although 3-D models / analytical tools are being developed for these special penetration areas, there is no current plan to correlate analysis to any test data!
- Penetration flow and understanding response of the thermal barrier is a very complicated scenario





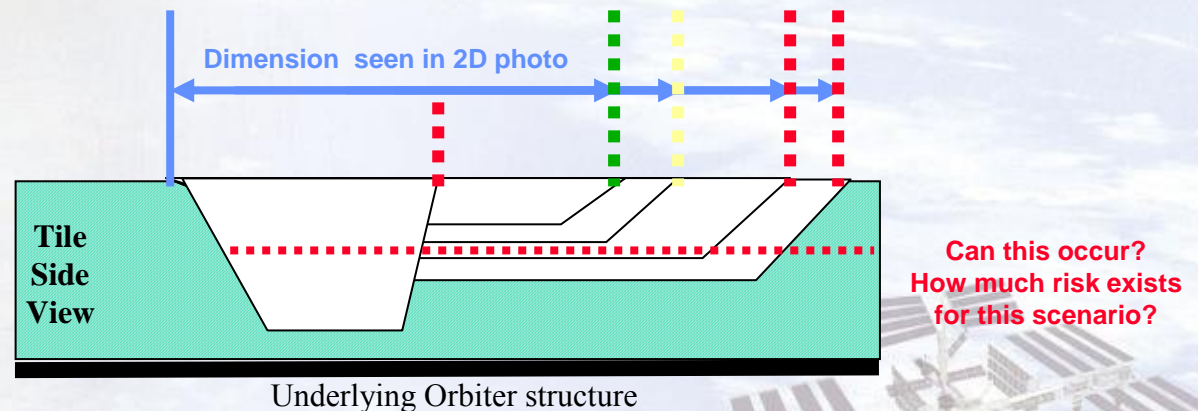
# Risk of “missing something with only 2D inspection” versus Ops Trade-space result unknown at this time!

## “Standard Gouge”



## “Deep Penetration”

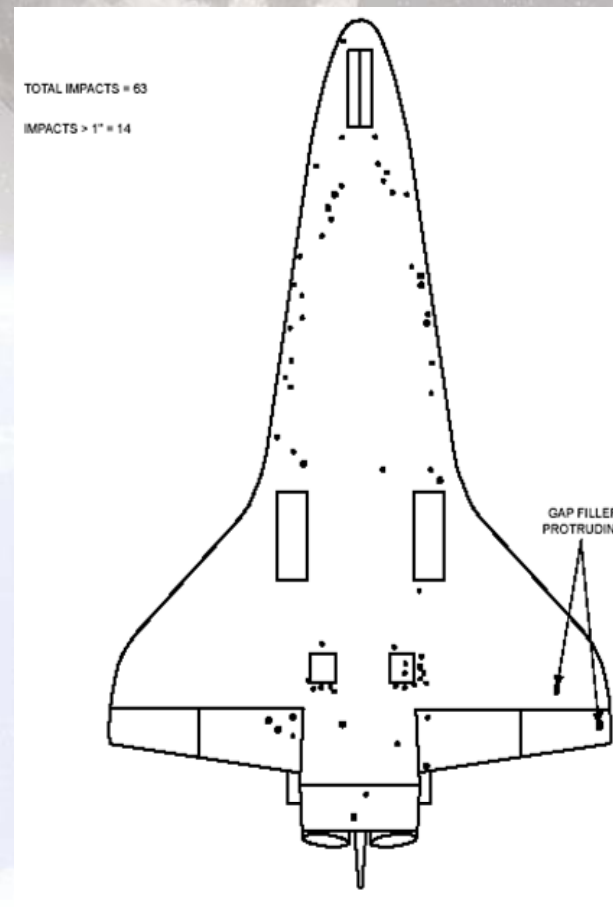
*Protecting for this could seriously affects OBSS activities and ops!*





# OV-102 Flight Damage History

Mission		Impacts > 1"	Total Impacts
STS-9	6	14	58
STS-61C	7	39	193
STS-28R	8	20	76
STS-32R	9	15	120
STS-35	10	17	147
STS-40	11	25	197
STS-50	12	45	184
STS-52	13	16	290
STS-55	14	13	143
STS-58	15	26	155
STS-61	16	16	97
STS-65	17	21	151
STS-73	18	26	147
STS-75	19	17	96
STS-78	20	12	85
STS-80	21	8	93
STS-83	22	13	81
STS-94	23	12	90
STS-87	24	132	308
STS-90	25	20	131
STS-93	26	49	208
STS-109	27	18	98
OV-102 Average		26.1	143.1
Fleet Average		30.5	144.9



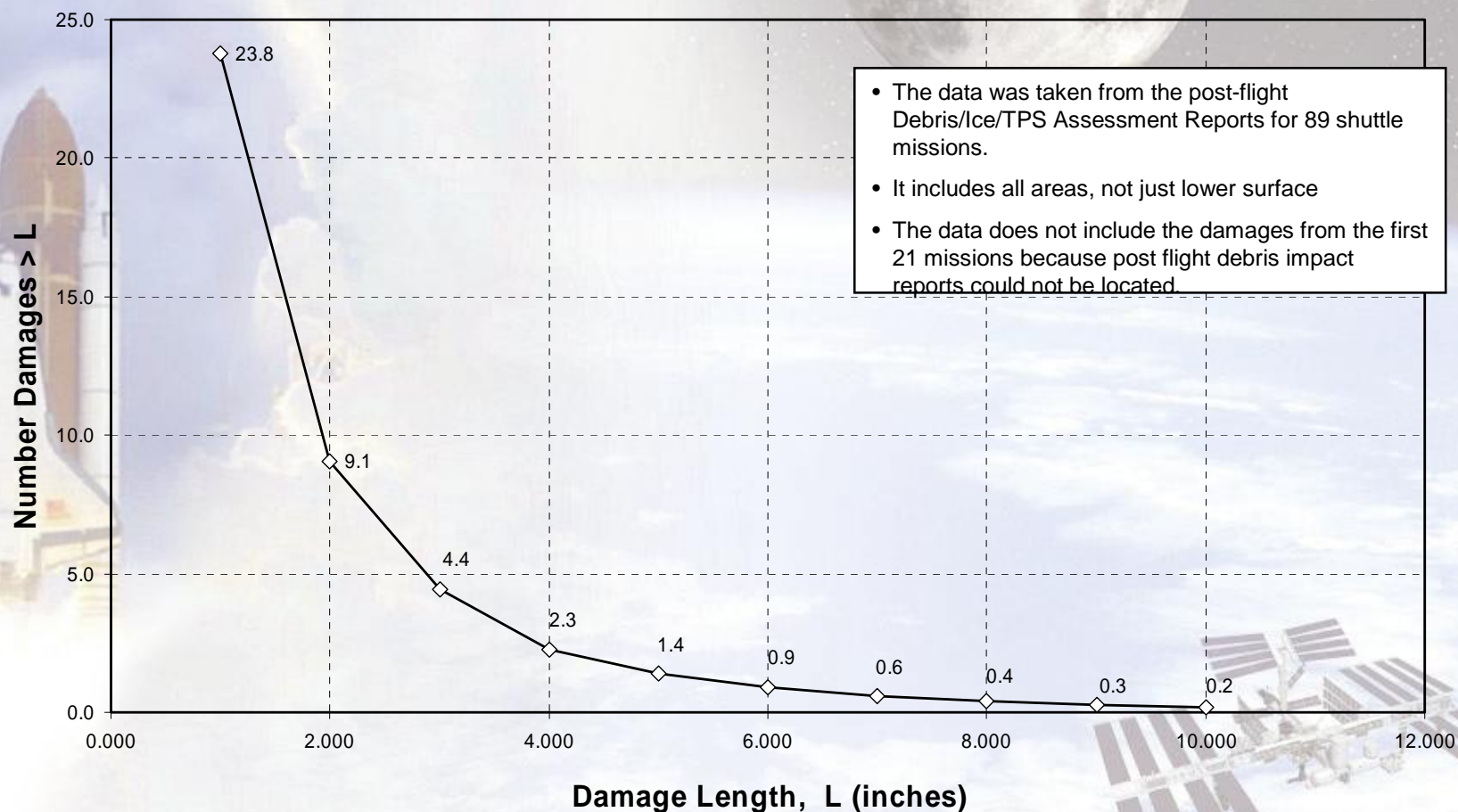
STS-109 Lower Surface Impact Damage





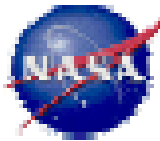
# Flight Damage History

## Average Number of Impact Damages Exceeding Length L per Flight





# Historical Tile Damage Background



- Orbiter has sustained greater than ~15,000 tile damage events (of varying degree) throughout life of Program
- Per knowledgeable TPS technical community, only "a few" damage sites would have been candidates for even considering an on-orbit repair, had that option been available, based on ground inspection post-flight
- Our flight history tells us that tile gets damaged during ascent on every flight
  - Judgment would also say that the modifications to the other elements will not preclude our tile damage "flight history" from being generally repeated on future flights (although some possible improvement against "big" damages is predicted)
- Our flight history tells us that the vehicle is robust to enter with the tile damage suffered to date for the particular mission conditions experienced
- Two potential "really tough" scenarios brewing:
  - Pre-flight risk: TRP, solely using TRP delivered use-as-is analytical tools, is only able to validate a very small inspection criteria and, based on our flight history, drive a recommendation toward a very ops intensive / timeline impacting OBSS inspection process per flight. (an inspection criteria that just doesn't "feel right" based on our gut)
  - Real-time risk: Real-time team, solely using TRP delivered use-as-is analytical tools, recommends performing high-risk repairs at a high rate of frequency (a rate that just doesn't "feel right" based on our gut)... but has nothing else to provide any technical rationale to stand behind.



# Pre-Flight Risk Assessment Philosophical Approach

## Raw Data Activity, Creating the RAIV data set

### Data Mining/Formatting

"Retro-actively" apply the tile damage inspection criteria

**(3" for acreage, 1" around door seals)**  
to previous flight history capturing violations per flight and per PRACA zone

**Note:** No available information for STS-41B & STS-41D, STS-1 through STS-5 eliminated from data set due to old and significantly different configs we were not interested in capturing, other major excursion flights (STS-27R, STS-87) to be discussed in more detail later.

**Any "Big Damage" trends seen along the way?**

## Technical Judgment

### TPS PRT Review

Review all inspection criteria violations and provide a judgment as to which of the violations should be considered "close calls"

### TPS PRT Review

Review "close calls" and provide a judgment as to whether "close calls" should be filtered out of data (i.e., not ascent debris, confidently corrected and verified debris source, etc.)

### TPS PRT Review

Review "close calls" and provide a judgment as to whether any other "forward looking" augmentation factors should be applied

### Result: "Residual Risk"

## Statistical Activity

### Statistical "Crunching"

Using flight history data and "residual risk", perform assessment to determine:

- 1) Likelihood of OBSS inspection requirement
- 2) Likelihood of "close call" damage





# Pre-Flight Risk Assessment Observations, Results, & Conclusions

		CASE1					CASE2					CASE3					
Region		Total Hits	Percent of Total	Laplace Score	Mean	95th	Total Hits	Percent of Total	Laplace Score	Mean	95th	Total Hits	Percent of Total	Laplace Score	Mean	95th	
Class Totals	Vehicle Total	549	100.0%	-8.5	5.3	17.4	175	100.0%	-2.1	3.5	8.6	150	100.0%	-2.4	3.0	7.1	OBJ2
	Lower Surface Tile Total	431	78.5%	-7.5	4.2	16.4	137	78.3%	-1.8	2.7	7.6	121	80.7%	-2.0	2.4	6.0	
	Generic Acreage Subtotal	189	34.4%	-4.3	1.8	6.9	70	40.0%	-0.1	1.4	4.0	66	44.0%	-0.2	1.3	4.0	
	Wing Glove Subtotal	60	10.9%	-2.8	0.6	1.0	25	14.3%	1.3	0.5	1.0	12	8.0%	1.8	0.2	1.0	
	Aero Surfaces Subtotal	37	6.7%	-0.1	0.4	1.0	16	9.1%	0.0	0.3	1.0	17	11.3%	-0.3	0.3	1.0	
	Special Penetration Areas Subtotal	145	26.4%	-6.2	1.4	5.9	26	14.9%	-5.3	0.5	3.0	26	17.3%	-5.0	0.5	3.0	
	No Zone ID Subtotal	47	8.6%	-7.1	0.5	2.0	0	0.0%	Sparse	Sparse	Sparse	0	0.0%	Sparse	Sparse	Sparse	
	Upper Surface Tile Total	71	12.9%	0.8	0.7	3.0	38	21.7%	-1.0	0.8	2.6	29	19.3%	-1.4	0.6	2.0	
	Wing Glove Right	35	6.4%	-4.2	0.3	1.0	10	5.7%	0.0	0.2	1.0	5	3.3%	0.0	0.1	1.0	
	Wing Glove Left	25	4.6%	0.6	0.2	1.0	15	8.6%	1.7	0.3	1.0	7	4.7%	2.4	0.1	1.0	
	Generic Acreage Right	70	12.8%	-1.4	0.7	3.0	28	16.0%	-0.9	0.6	2.6	27	18.0%	-0.7	0.5	2.6	
	Generic Acreage Left	86	15.7%	-5.0	0.8	4.9	25	14.3%	1.8	0.5	2.6	22	14.7%	1.9	0.4	2.0	
	Wing and Acreage Right	105	19.1%	-3.6	1.0	3.0	38	21.7%	-0.7	0.8	3.0	32	21.3%	-0.6	0.6	2.6	
	Wing and Acreage Left	111	20.2%	-4.1	1.1	6.0	40	22.9%	2.5	0.8	2.6	29	19.3%	2.8	0.6	2.0	

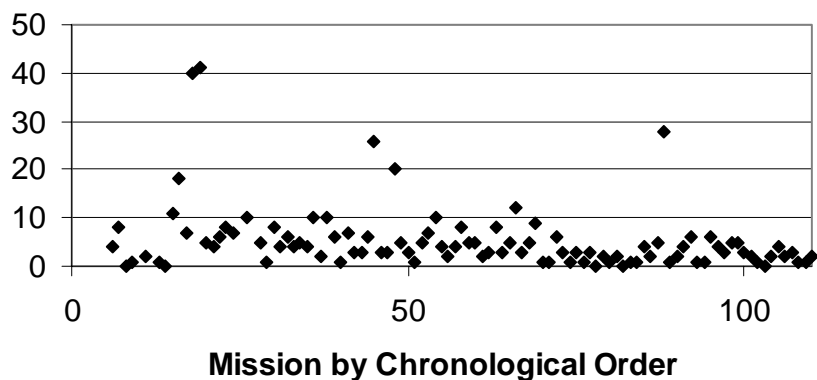
Legend	
CASE1 =	Total RAV data set (103 missions), excluding STS-1 thru STS-5 and STS-27R
CASE2 =	RAV data set for the last 50 missions only
CASE3 =	RAV data set for the last 50 missions only, excluding STS-87

\* Green denotes a decreasing trend, red denotes an increasing trend

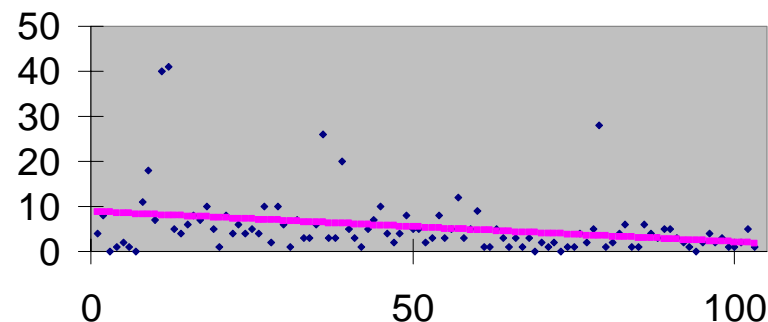


# Pre-Flight Risk Assessment: Observations, Results, & Conclusions

**Total by Mission  
STS-27R Removed**



**Associated Regression**



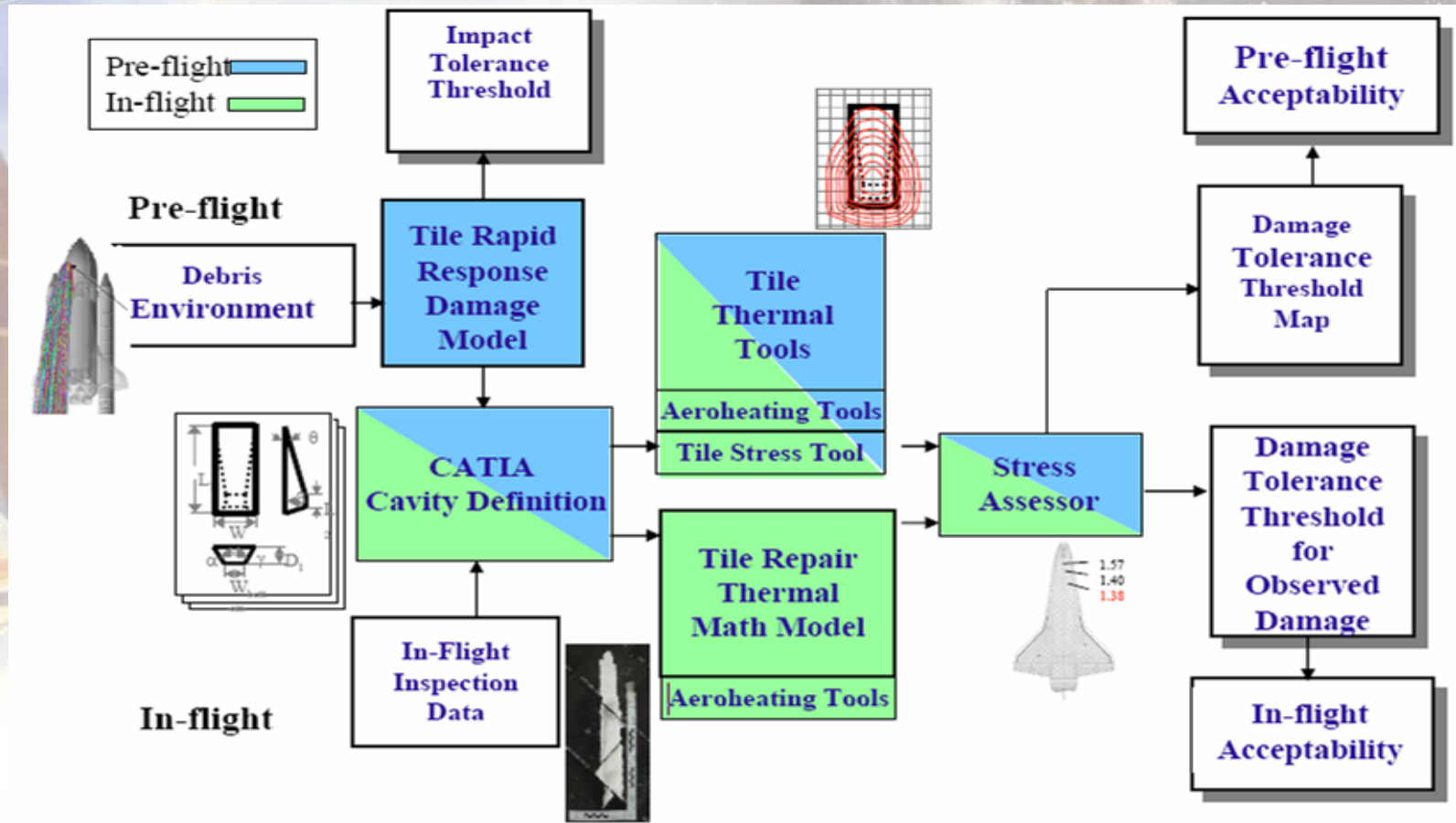
These graphs portray the total significant hits by mission ordered chronologically, less STS 1- 5 and 27R. Evident from both graphs is the general downward trend in total number of significant hits with a greater degree of variability in the first 50 as compared with the last 50. This is indicative of a distribution that, over time, has a decreasing mean and variance. This is similar to a production process that has increasing control and a lowering set point.

**Legend**

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# Tile Models to Determine Impact and Damage Tolerance Thresholds



ORB-228

George K. Gafka 281-483-7732





# RCC and Tile Tools and Models

Models	New / Updated / Existing	Used For Pre- Flight C/E	USED REAL TIME			
			Launch Go/No- Go	On-Orbit before Inspection	On-Orbit Use-as-Is	On-Orbit Repair
<b>RCC Damage Prediction Tools</b>						
LESS Dyna Tool	N	X		X		
Rapid Response RCC Damage Prediction Tool	N		X	X		
<b>RCC Aeroheating Tools</b>						
Step/Ramp Heating	N					X
LESS Breech Internal Flow Model	N				X	
RCC Damage Growth Tool	N				X	
<b>RCC Thermal Models</b>						
RCC 3D Thermal Math Models	E				X	X
<b>Tile Damage Prediction Tools</b>						
Tile Rapid Response Damage Model (foam)	N	X	X	X		
Tile Rapid Response Damage Model (ice)	N	X	X	X		
Tile Screening Tool	N	X				
<b>Tile Aeroheating Tools</b>						
Cavity Heating Database	N	X			X	
CFD for Cavity Heating	N	X			X	
Catalytic Heating Tool	N	X			X	
Boundary Layer Transition Prediction Tool	N	X			X	X
<b>Tile Thermal Tools</b>						
2D Thermal Model	N				X	
3D Acreage Tile Thermal Model	N	X			X	X
Repaired Tile Thermal Model	N					X
Special Configuration Thermal Models	N				X	
<b>Tile Stress Tools</b>						
Tile Stress Tool	N	X			X	X
Tile Bondline Integrity Tool	U	X			X	X
Stress Assessor Tool	N	X			X	X

ORB-230



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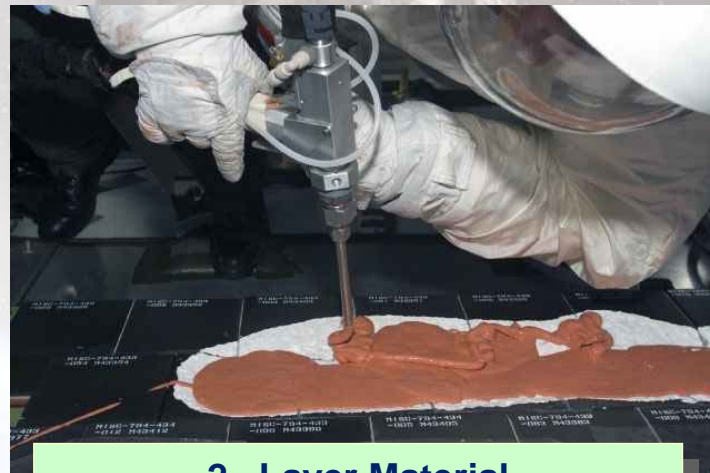
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# Repair Procedure Overview



**1. Trim Gap Filler as Required  
Clean Tile with Gel Brushes**



**2. Layer Material**



**3. Flatten / Smooth Repair**





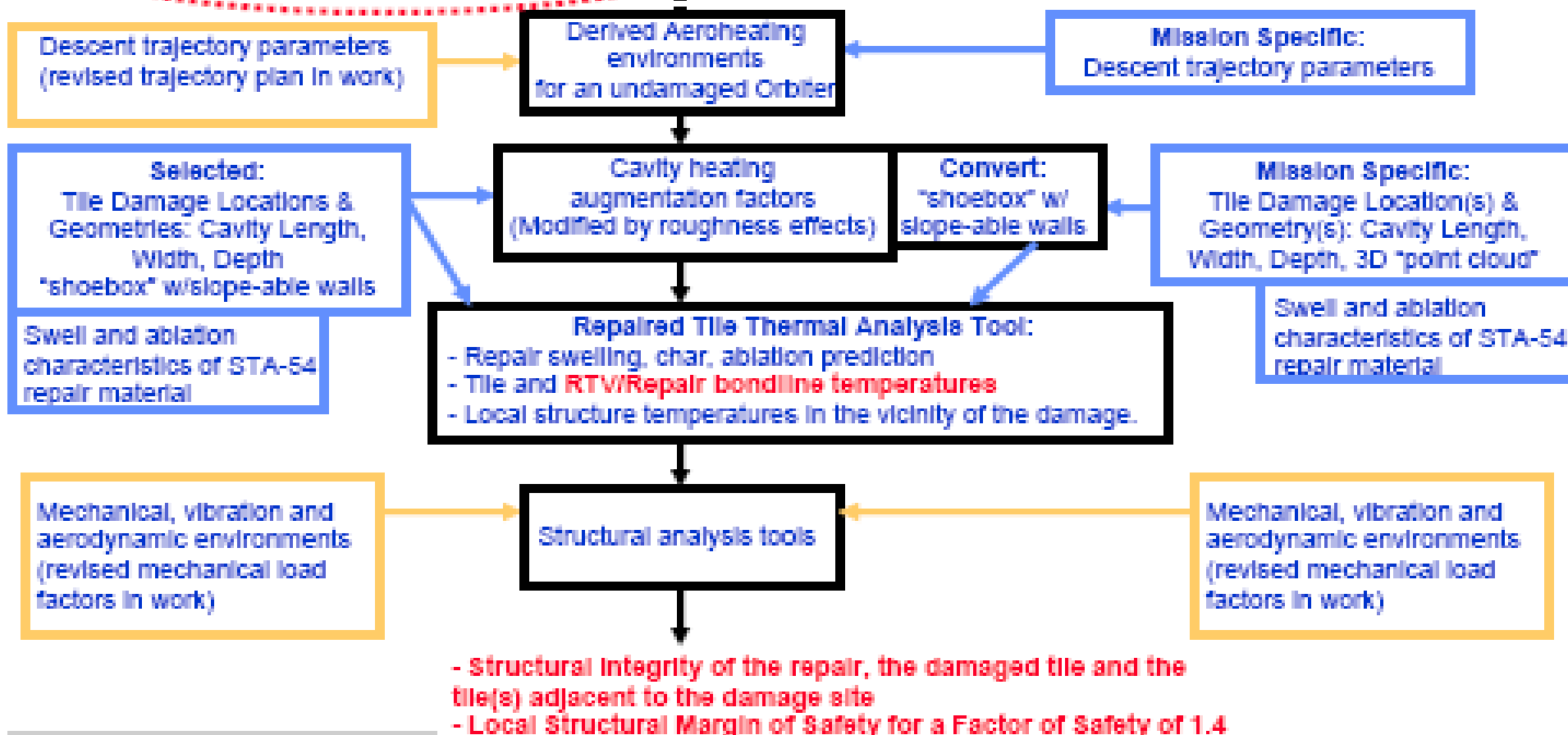


# Development of TRP Repair Disposition Analytical Tools



## Pre-Flight Development

## Real-Time Mission Specific Use

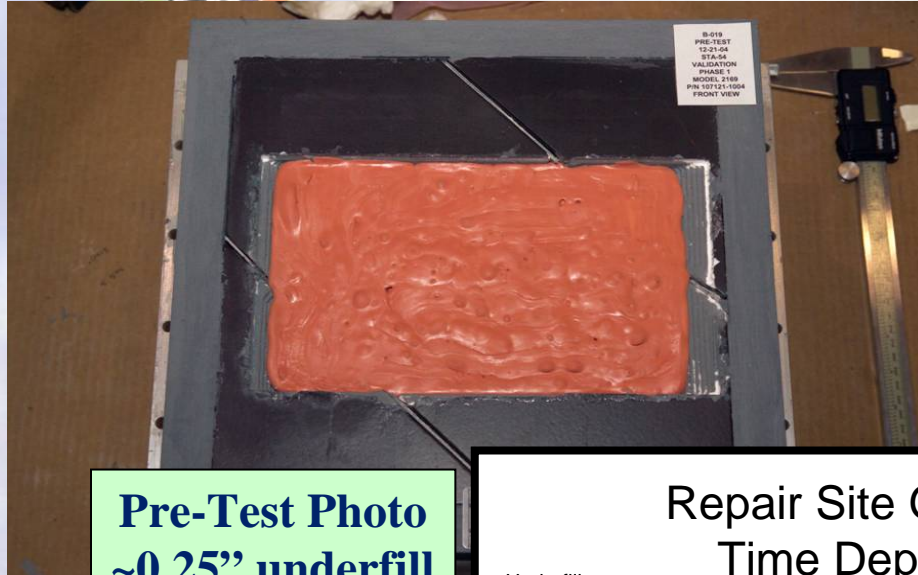


### Legend

Analysis Tools Data Result



# Test Article Exposed to Low Shear Test Condition Model #2169 – 9"x5" Cavity Filled in HTV 2

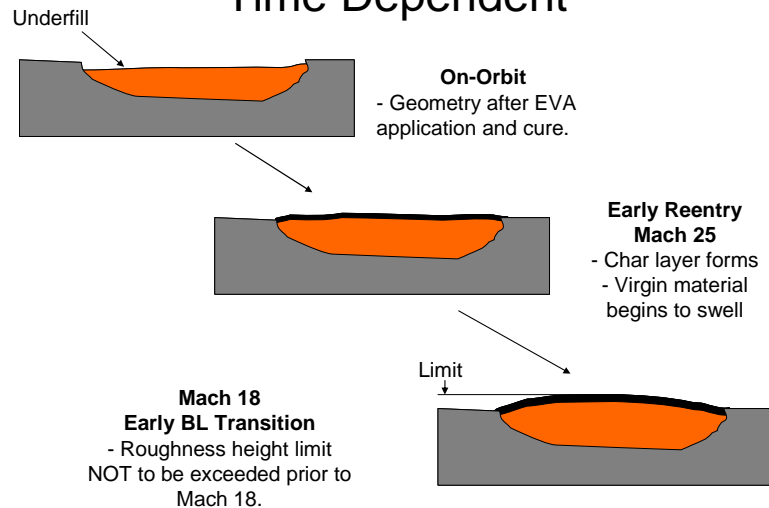


**Pre-Test Photo**  
**~0.25" underfill**



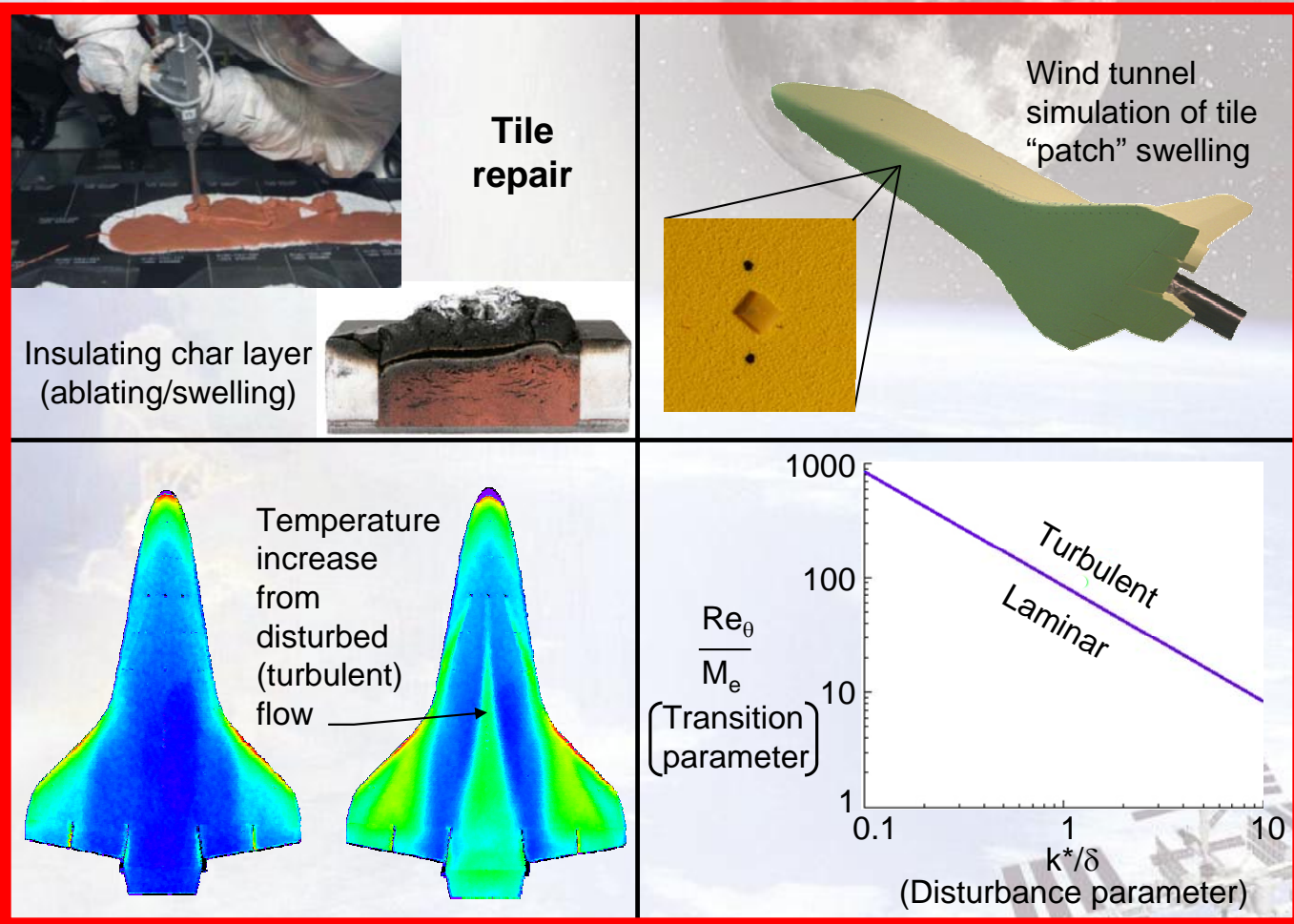
**Post-Test Photo**  
**~0.25" swell above tile**

## Repair Site Geometry Time Dependent





# Development of Transition Prediction Methodology



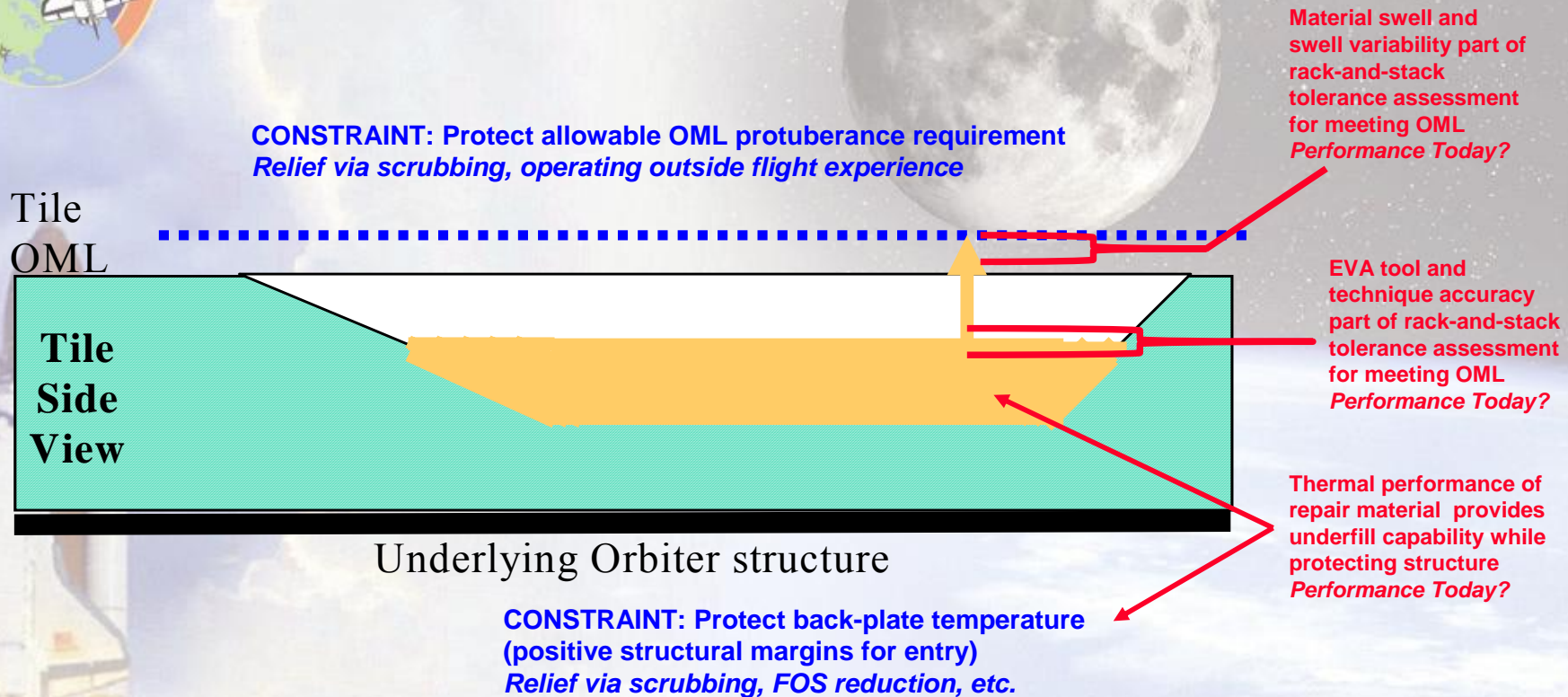




# LOCAL DAMAGE SITE

## Trade-space result unknown at this time!

### RESULT: Possible Capability Black-Out Zones



**Note: There is also a “global” or downstream effect that must be considered. This can result in additional blackout zones if “low margin” healthy or damaged downstream tiles see elevated temperatures that would result in the underlying structure temperature exceeding allowable limits.**  
*Relief via scrubbing, FOS reduction, etc.*

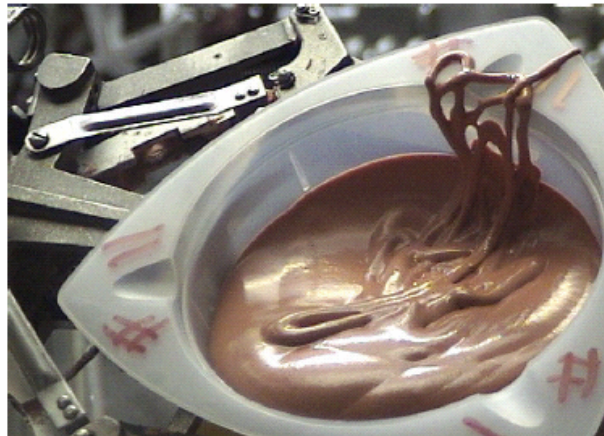


# Killer/"Golden" Requirements

## Thou shall have NO bubbles...

- Initial sample, Part A - CIPAA 1005

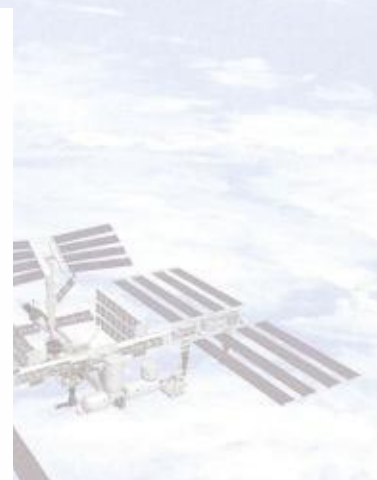
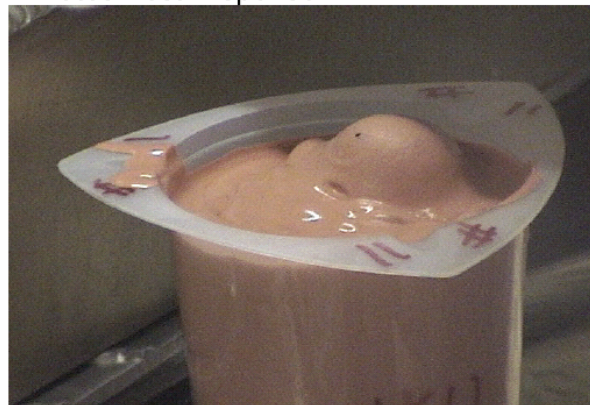
Following Dispense



15 minutes Post-Dispense



- Initial sample, Part A - CIPAA 1005
- 30 minutes Post-Dispense

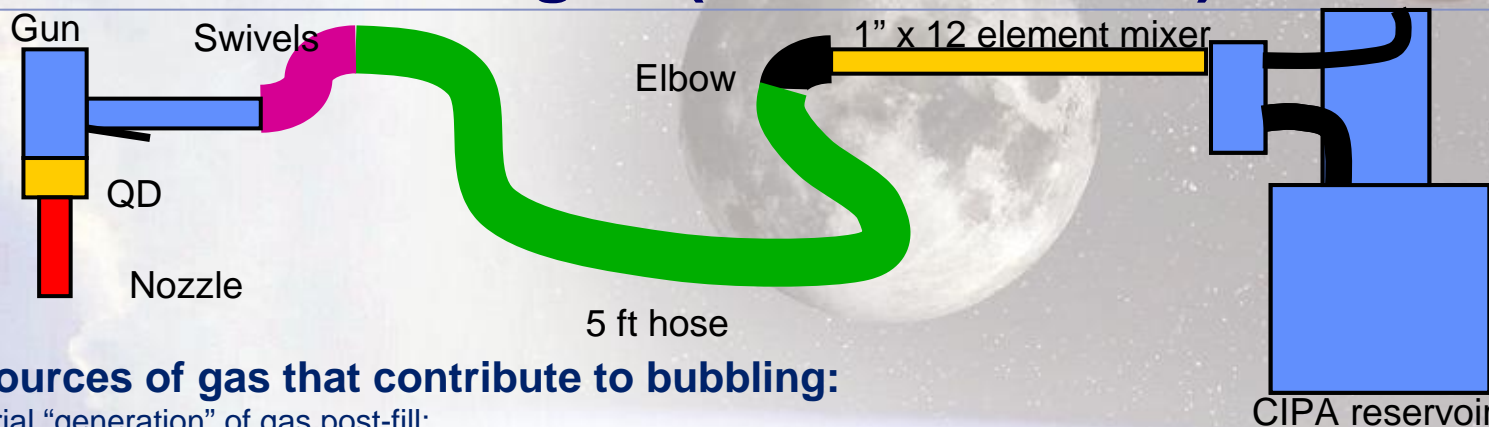






# Example of Hardware/Test Configuration

## Sources of gas (5 sources?!?)



### Five possible sources of gas that contribute to bubbling:

- Internal-to-the-material “generation” of gas post-fill:
  - 1 • Residual gas remaining in material (Part A) post degassing
    - Resulting gas could nucleate into bubbles over time, could be “pulled out” of solution with pressure drop (cavitation)
    - **Data suggests likely contributor, can’t fully exonerate or confirm**
  - 2 • Micro-balloons breaking post degassing
    - Resulting gas could nucleate into bubbles over time, could be “pulled out” of solution with pressure drop (cavitation)
    - **Analysis suggests extremely sensitive to number allowed to break, possible contributor, can’t fully exonerate or confirm**
  - 3 • Ethanol???
- External-to-the-material influences “feeding” the material gas:
  - 4 • Ambient air leaking past environmental seal during storage
    - Could nucleate into bubbles over time, could be “pulled out” of solution with pressure drop (cavitation)
    - **Data suggests likely contributor, can’t fully exonerate or confirm**
  - 5 • Nitrogen pad pressure leaking past dynamic seal during system pressurization
    - Could nucleate into bubbles over time, could be “pulled out” of solution with pressure drop (cavitation)
    - **Data suggests NOT a likely contributor, can’t fully exonerate or confirm**

**Conclusion: No way to fully preclude bubbling with this material/hardware system!**  
**So, instead how sensitive is system/entry performance to bubbles?**





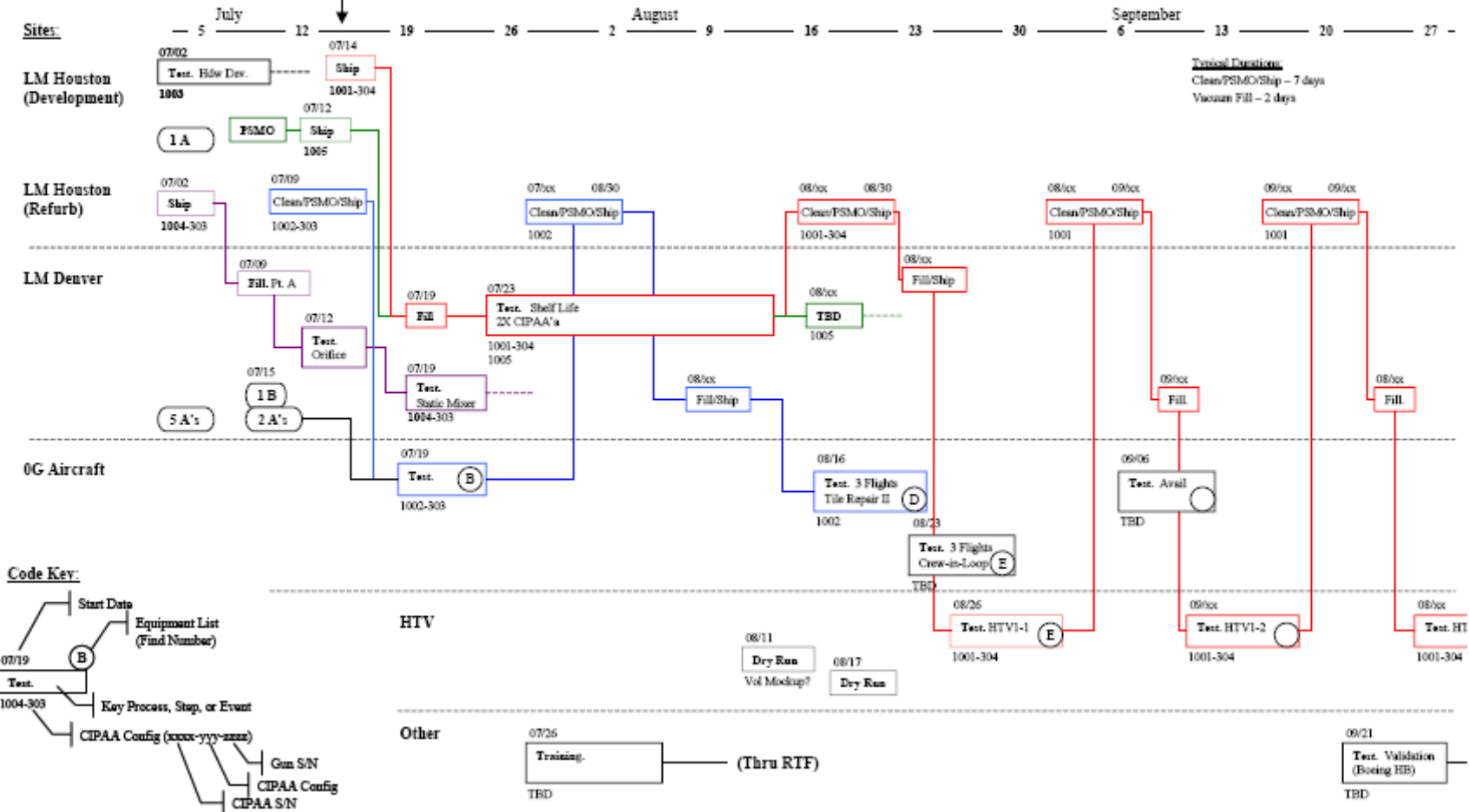
# Logistics Deployment Chart Near Term Planning Tool

## Tile Repair Project – Major Material and Equipment Logistics Deployment Chart (AKA “Swim-lane Chart”)

1. CIPAA Units
2. Transfer Containers
3. CIPAA Gusa

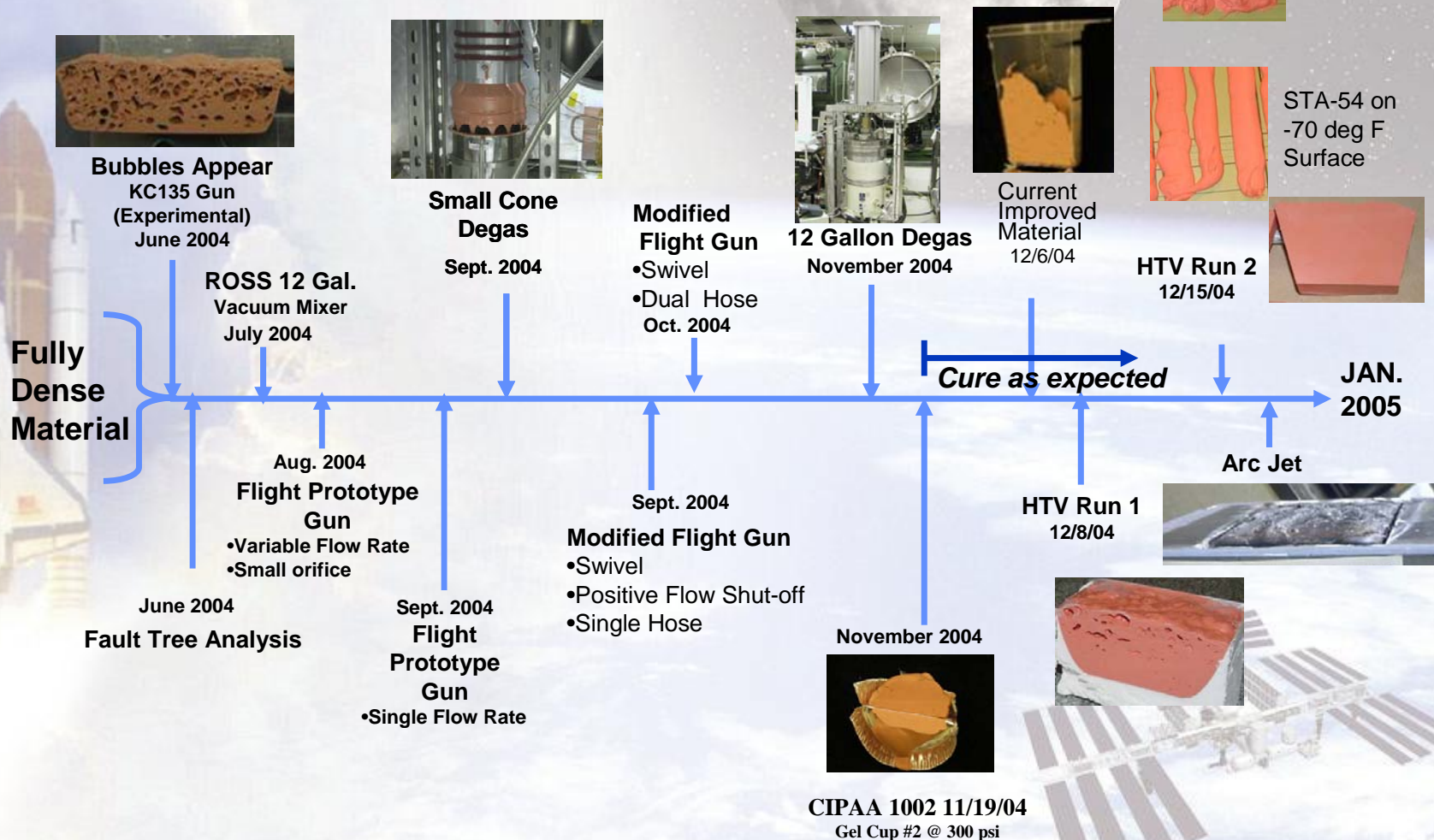
**3 Months (Rolling)**

Updated: 7/14/2004





## Tile Repair – STA-54 Material / Hardware Process Improvements and Test Timeline





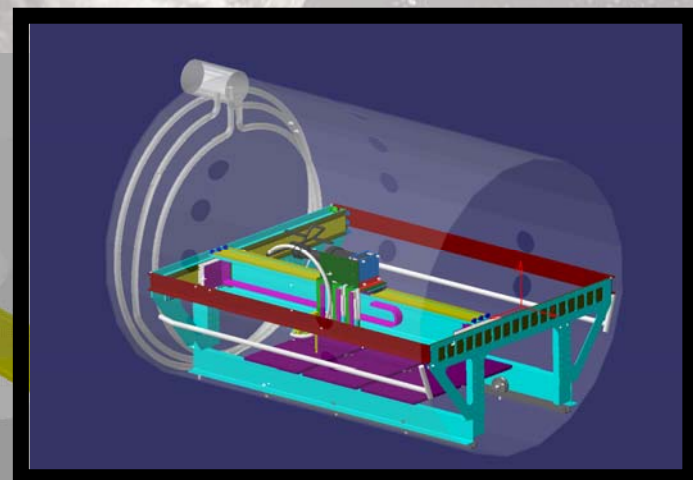
# Repair Ground Test Equipment Gantry System Configuration

**C-3 Chamber  
Interior Outline**

**CIPAA unit**

**24"x24" Damaged  
Tile Arrays**

**X-Y-Z Linear Motors**





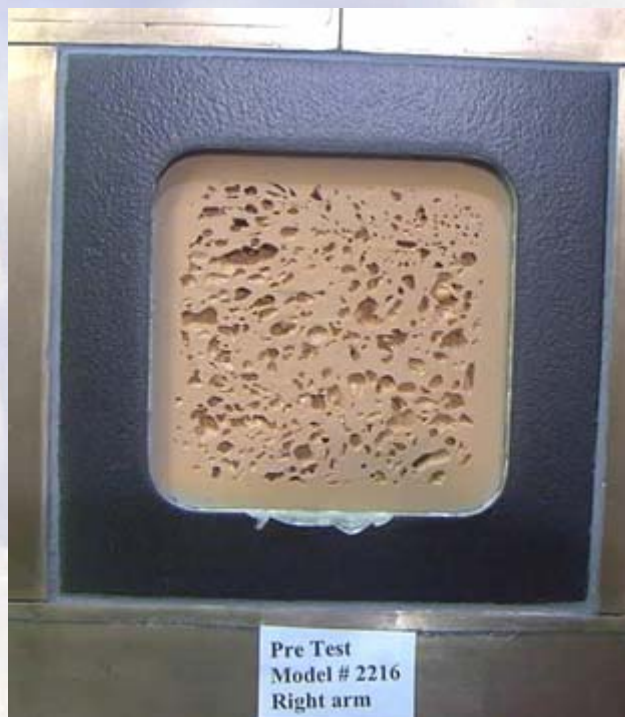


# STA-54 VOID EFFECTS TEST PROGRAM

## MODEL #2216 PRE AND POST TEST PHOTOS

### 0.25 INCH UNDERFILL

### COMPRISED OF THREE 0.50 INCH THICK LAYERS





# Observations, Ideas, and Opinions Presentation Outline

- **Project Management & Systems Engineering Challenges**
  - In The Beginning...What is your mission? Can you “certify” to it?
  - Team Roles/Responsibilities/Requirements/Contracts/Deliverables
  - Use-As-Is becomes most critical capability!
    - Flight History Database, a surprisingly contentious topic
  - Tile Repair is really tough, becomes “best effort” for RTF
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    - Tough Trade Spaces
  - Delivery for RTF
- **STS-114**
- **Conclusion**

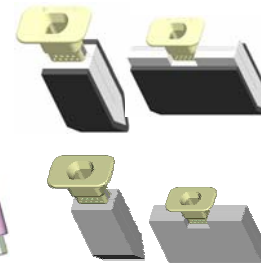
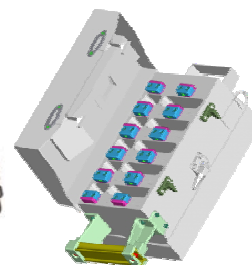
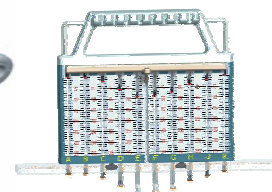
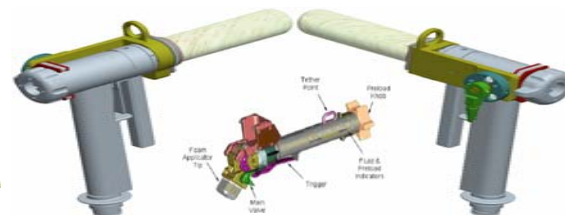
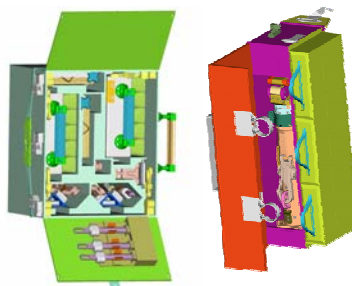
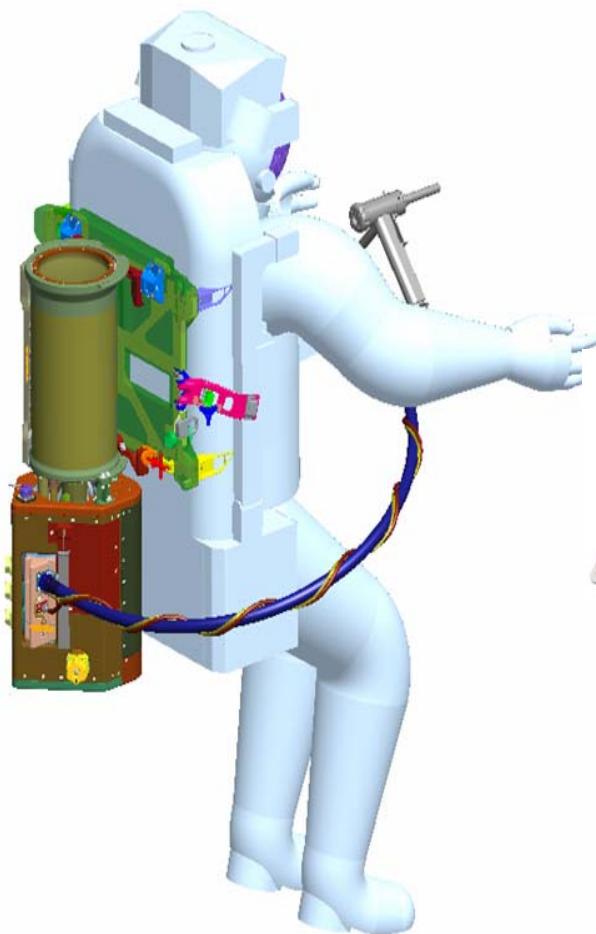
You are here.

- **Understanding/Influencing/Accepting Your Environment**
  - Cost, schedule, technical/safety, political, emotional
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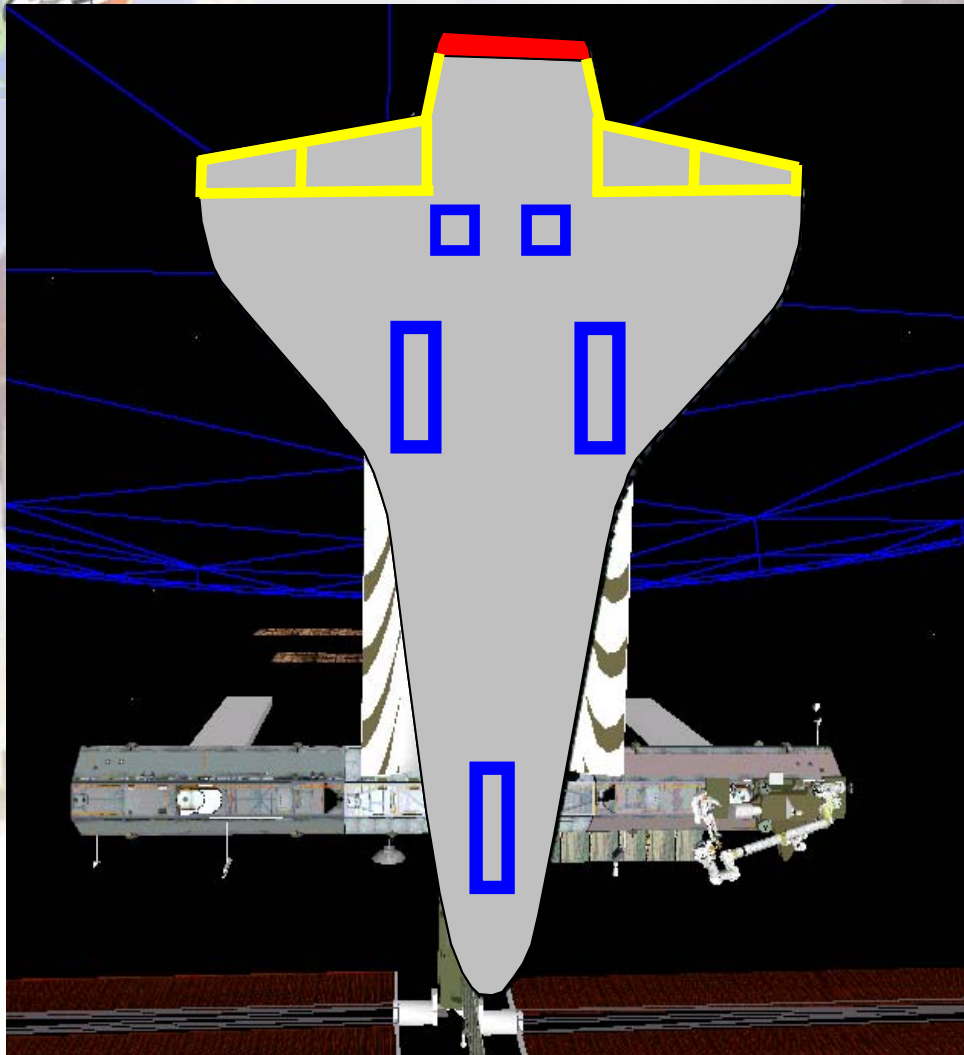
# Tile Repair Hardware Suite







# Tile Repair Project – A View of Project Scope



## RTF



Door Seals



Acreage

## Future



No Access



Challenging Geometry



# Tile Repair Project – A View of Project Scope

TPS Area	Likelihood of Damage	Conseq. of Damage <b>*TBR</b>	Detect-ability	Current EVA Access-ability	Current Design Appr Compatible w/damage?	RTF Support-ability	Required for RTF (TRP opinion)
Acreage Tile (Lower Surface)	?	H > 3"	Yes	Yes	Yes	H	Yes
Chine/Wing Glove	?	H > 3"	Yes	Yes	At risk	L	Yes/No?
Door Seals	?	H > 1"	Yes	Yes	At risk	M	Yes
LESS Carrier Panels	?	H > 1"	Yes	1 – 20, Yes	At risk	M	Yes
				Outboard, No	At risk	L	No
Elevon	?	H > 3"	Acreage Only	Acreage, Yes	Yes	Acreage, H	Yes
		H > 1"		Other, No	Hinge, At Risk	Hinge, L	No
Vertical Tail	?	H > 3"	No	No	At risk	L	No
OMS Pod Tile	?	H > 3"	Not Inspected	Forward edge only	Accessible Acreage only	Acreage, H	Yes
				Other, No	Other, No	Other, L	No
Body Flap	?	H > 3"	Acreage Only	Forward acreage, Yes	Acreage, Yes	Acreage, H	Yes
				Other, No	Other, No	Other, L	No

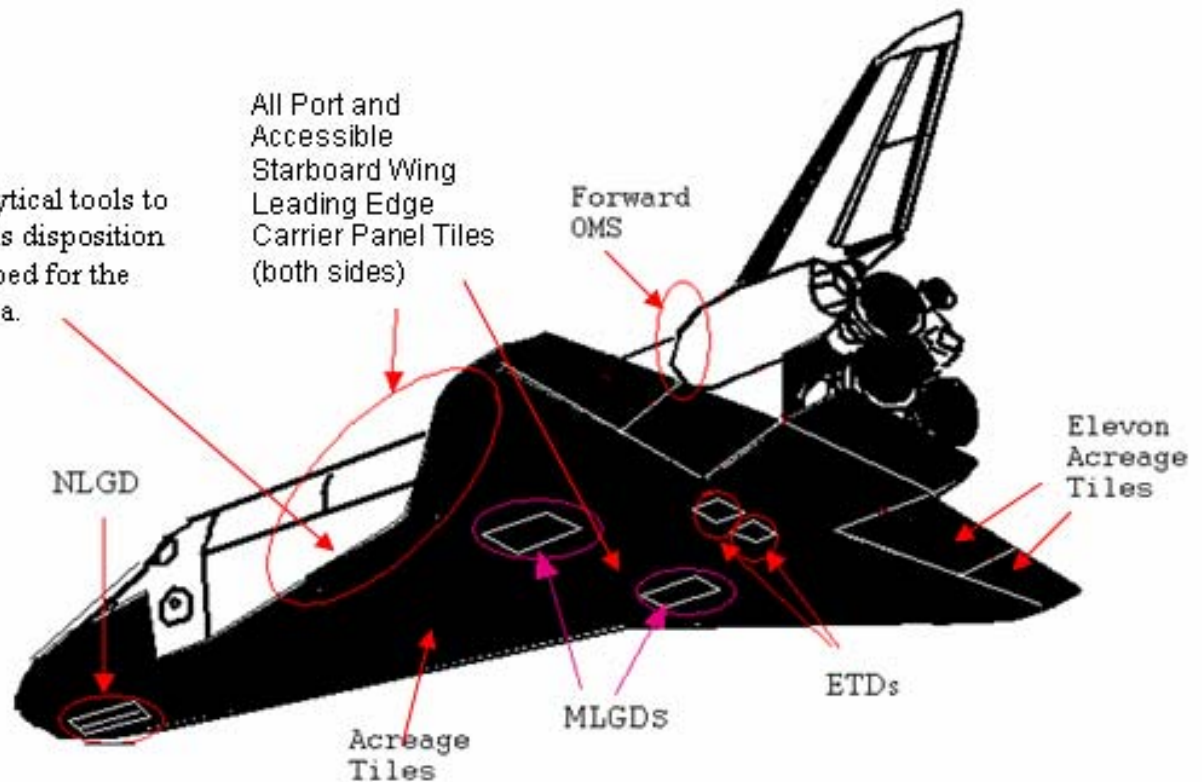


# System Requirements for RTF

Figure 3.2-1 Tile Damage Assessment and Repair Locations

Wing Glove

Note: only analytical tools to support use-as-is disposition shall be developed for the Wing Glove area.

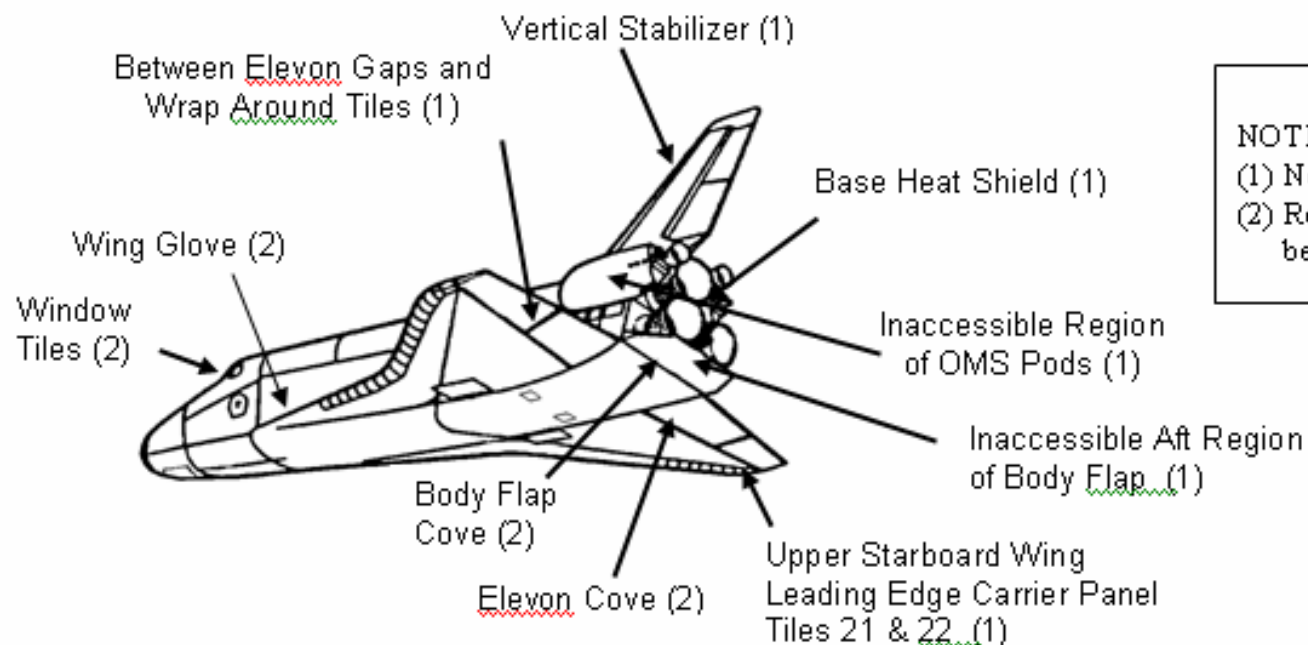






# System Requirements for RTF

**Figure 3.2-2 Examples of Tile Locations Not Explicitly Repairable by TRP**



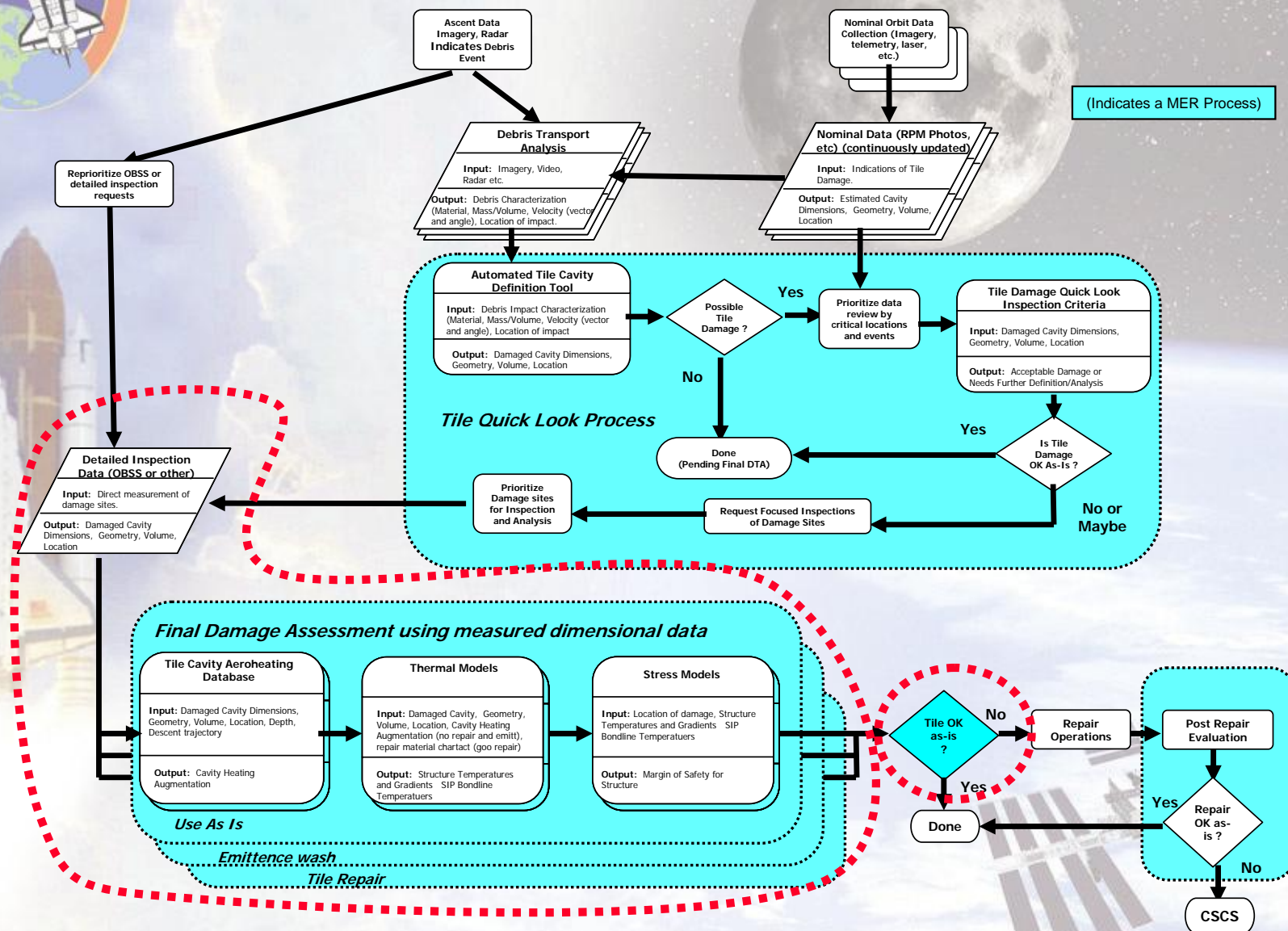
**NOTES:**

- (1) No EVA Access
- (2) Repair Configuration Cannot be Standardized or Analyzed



# Real-Time

## Tile Damage Assessment Process

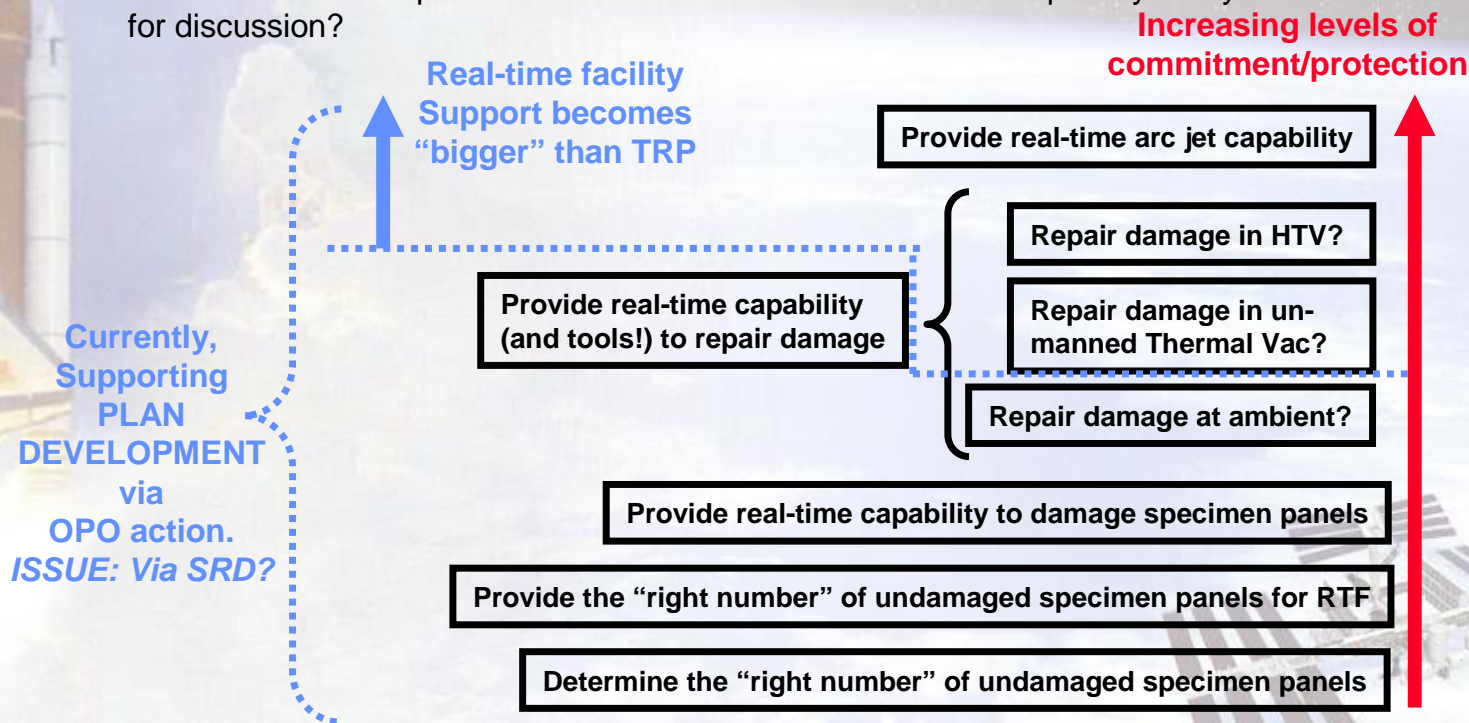




# Real-time Ground Test Capability (HTV, arc jet, etc.) for mission-specific damage/repair

## ●OPO/Program Direction

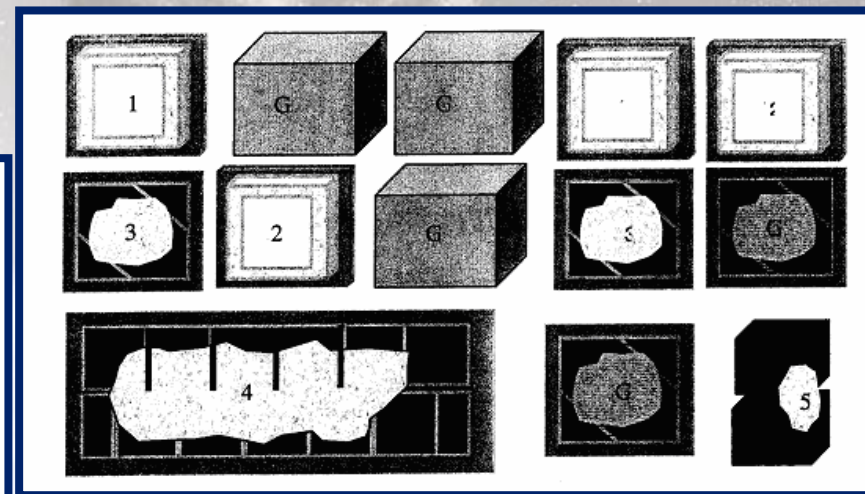
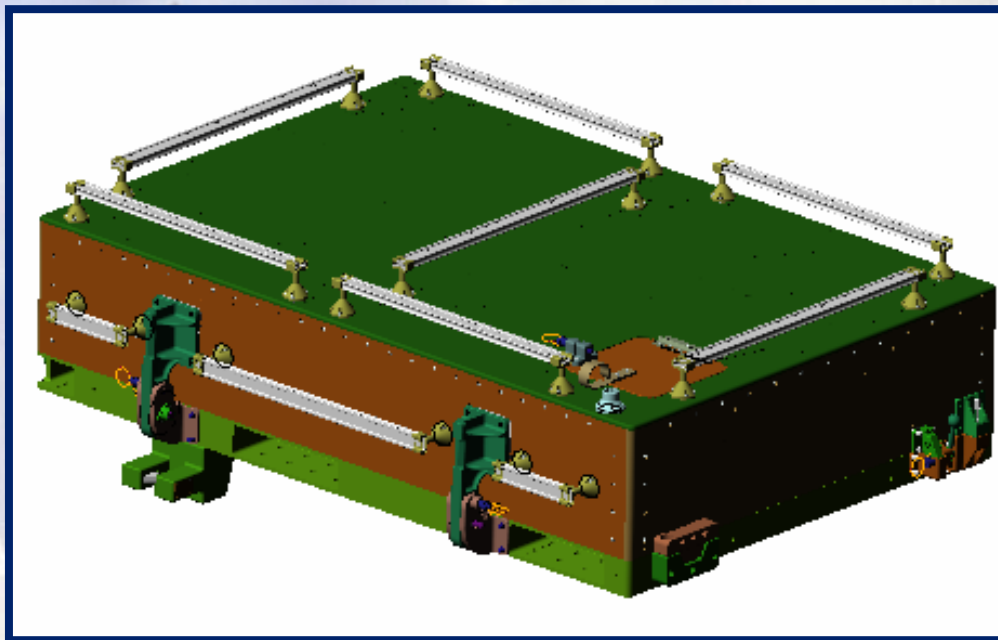
- Should TRP SRD contain requirements for providing deliverables and damage capability or continue to work to OPO action?
  - » Envisioned to be a part of nominal mission capability or short-term requirement for first few flights?
- What is the forward plan to take the “Real-Time Ground Test Capability” story forward to the Program for discussion?







# Thermal Protection System (TPS) Repair Development Test Objective (DTO)





# Tile Repair Project Conclusion

- **Use-As-Is Analytical Tools**

- Rigorously developed, test anchored, peer reviewed, documented, “simmed” and “certified” in support of Return To Flight (STS-114)
- Required and used successfully during STS-114 mission

- **Historical Database**

- Supplemental tool developed/delivered in support of Return To Flight (STS-114)
- Used as a sanity check for use-as-is predictions pre-flight
- Used successfully during STS-114 mission as a supplement to damage disposition activities

- **Tile Repair Capability**

- Best effort delivered and flew on STS-114
- Safe to fly, safe to use, system level functional performance for repair not certified, best data to date available for assessment
- Further CIPAA (“goo-based”) development recently canceled with continued support of other repair capabilities



# Observations, Ideas, and Opinions Presentation Outline

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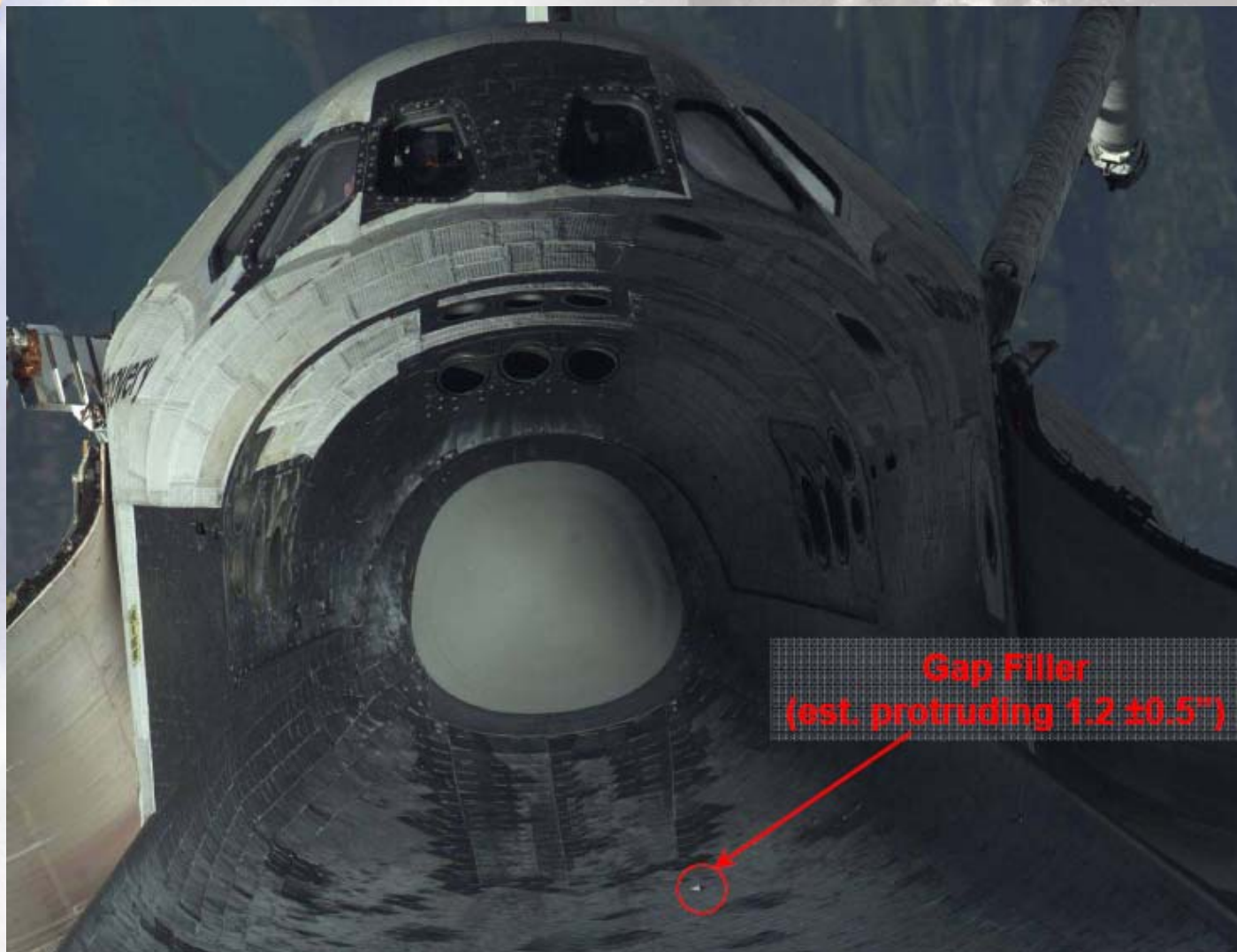
You are here.

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# STS-114... Flight Day 3, RPM “Quick Look”





# Use-As-Is Risk Summary

## KEY ASSUMPTION

1. BLT, Mach ~ 18

Current,  
"best estimate"

2. BLT, Mach 21.5

3. BLT, Mach 24

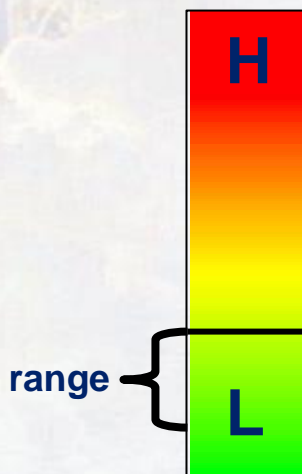
## UNCERTAINTIES AND SAFETY RISKS

Aero Heating: trajectory, BLT Mach number and heat rate/heat load

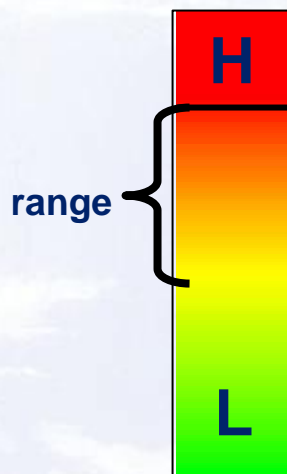
Thermal/Structural Analysis for specified case

Flight History support of analysis

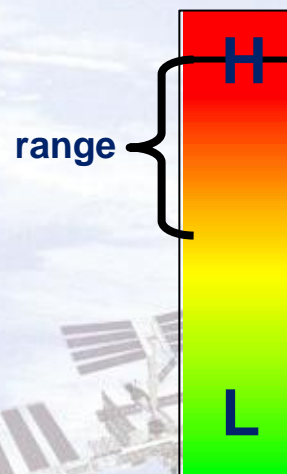
Flight Control Performance (Certified to Mach 19)



Minor Vehicle Damage  
Structural Integrity Maintained



Major Structural Damage /  
LOCV



Major Structural Damage /  
LOCV

## POTENTIAL CONSEQUENCES



# EVA Repair Risk Summary

## EVA 3 - Shuttle Airlock - SSRMS

**Expected outcome  
per KSC and TPS experts**

### REPAIR OPTION

1. Gap Filler  
Extraction - Finger

2. Gap Filler  
Extraction - Forceps

3. Hacksaw Cut

4. Scissors Cut

### SAFETY RISKS: COMMON

### SAFETY RISKS: UNIQUE

### Translation to/from Worksite and Inadvertent Damage

Inadvertent Damage  
Repair Confidence

Inadvertent Damage  
Repair Confidence

Inadvertent Damage  
Repair Confidence

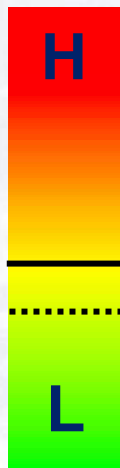
Contamination/Dust/FOD

Inadvertent Damage  
Repair Confidence

Contamination/Dust/FOD

(At the work-site)

COMMON .....→



### MISSION IMPACTS

For a nominal EVA 3, all primary Mission objectives can be accomplished (no significant impact). Unexpected/off-nominal EVA task durations may result in significant impacts. If the EVA is extended beyond 148 minutes, Mission impacts (additional EVA 4). 50





# STS-114 MMT

## Conclusions/Recommendations

### ♦ Recommend use-as-is disposition if, and only if:

- ♦ Confidence exists that on-orbit configuration represents Case 1 (BLT, Mach 18)
- ♦ NOTE: Likelihood appears low that we will get to here with confidence, especially in time frame that supports required MMT decision milestones
- ♦ NOTE: *This risk is driven solely by high uncertainties in key areas!*

**versus**

### ♦ Recommend repair attempt/disposition if:

- ♦ Confidence can not be established in the aero heating environments or vehicle response to those environments
- ♦ Case 2 (BLT, Mach 21.5) or Case 3 (BLT, Mach 24) is likely scenario
- ♦ Recommended repair order of implementation

♦ Try first: Gap Filler extraction – Finger

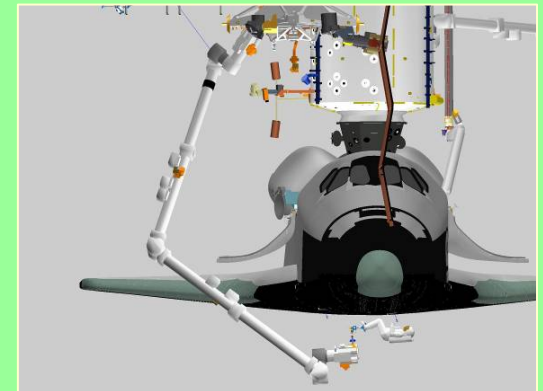
♦ Next: Gap Filler extraction – Forceps

♦ Next: Hacksaw

♦ Last resort: Scissors

♦ NOTE: *Consistent with current EVA plan*

♦ NOTE: *This risk is driven by consciously choosing to accept a, better understood and easier to control/manage (relative to use-as-is), risk*



Generic recommendation logic



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  - Healthy tension, good push back



# Typical “Peer Review” of Documentation

Required underfill dimension to account for swell at each location	<u>TPS Mission Central Website</u>	<u>Flight Control Team</u>	<u>Damage Assessment Team</u>	Prepare EVA Procedures	<del>Deleted: Crew console TPS PRT</del> <del>Deleted: Analysis PRT</del>
Analysis Report of damage sites.	<u>TPS Mission Central Website</u>	<u>Damage Assessment Team</u>	Analysis PRT	Used to prepare report specifying repair, or disposition	<del>Deleted: ¶</del> <del>Deleted: TPS PRT</del>
Presentation of dispositions	<u>MER MMT</u>	<u>OPO</u>	<u>Damage Assessment</u>	Used to turn on repair effort	<del>Deleted: OBO/ PIT/MMT</del> <del>Deleted: TPS PRT</del> <del>Formatted: Normal</del>

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# Presentation! Presentation! Presentation!

## Typical day at the Space Shuttle Program Requirements Control Board (SSPRCB)

CR/ACTION OPR TITLE/ACTION DESCRIPTION

S042013EV DELETE NITROGEN TANK AND AFT BALLAST BOX FROM JSC-MO STS 121, STS 300 AND STS 115 DEFER - 10/29/04 SSP PRCB PRESENTER(S): JSC-MO3-----

S050411AF SUBMITTAL OF DCN 041 TO HAZARD REPORT S.10, JSC-MX PARTIALLY OPEN GO2/GH2 VENT/ RELIEF VALVE INDICATED CLOSED DEFER - 10/29/04 SSP PRCB PRESENTER(S): MSFC-ET-----

S050430BG CHANGE TO BASELINE ORBITER HAZARD REPORT - JSC-MX ORBI 036 DEFER - 11/04/04 SSP PRCB PRESENTER(S): USH-0E ----

S050430BH CHANGE TO BASELINE ORBITER HAZARD REPORT - JSC-MX ORBI 256 DEFER - 11/04/04 SSP PRCB PRESENTER(S): USH-OE --

S060348 BASELINE SAFETY AND RELIABILITY REQUIREMENTS JSC-MO FOR SPACE SHUTTLE CARGO INTEGRATION HARDWARE DEFER - 10/29/04 SSP PRCB PRESENTER(S): JSC-MO2 -----

S062190A PROVISION FOR REPLACEMENT SPECIFICATIONS IN JSC-EA SHUTTLE PROGRAM M&P REQUIREMENTS DEFER - 10/29/04 SSP PRCB PRESENTER(S): JSC-ES4 -----

S062253 UPDATE TO SE-S-0073 SPECIFICATIONS FOR KSC-MK-SIO POTABLE WATER DEFER - 10/29/04 SSP PRCB PRESENTER(S): JSC-SF23 -----

S062292A UPDATES TO APPENDIX R, THE SPACE SHUTTLE KSC-MK PROGRAM CONTINGENCY ACTION PLAN DEFER - 10/29/04 SSP PRCB PRESENTER(S): KSC-MK-SIO -----

S062313 RETURN TO NIGHT LAUNCH OPPORTUNITIES JSC-MS JSC-MS/1-1 DEVELOP A PLAN TO DOCUMENT CRITERIA FOR RETURN TO NIGHT LAUNCH, INCLUDING OBJECTIVES WHICH MUST BE MET AND HOW OBJECTIVES ARE MET FOR DAY LAUNCHES AND NIGHT LAUNCHES. REPORT TO THE PRCB. DEFER - 11/18/04 SSP PRCB PRESENTER(S): TBD -----

S062343 ACTIONS ASSIGNED FROM THE JUNE 9, 2004 SPACE JSC-MS FLIGHT LEADERSHIP COUNCIL JSC-MS/2-1 USING PREVIOUS ORB FLT HISTORY, DEVELOP & VALI- JSC-MV/2-2 DATE CRITERIA FOR DETERMINING WHEN DISPOSITION OF DAMAGE OR SUSPECTED DAMAGE TO THE ORB TPS REQUIRES ADDITIONAL, HIGHER RESOLUTION, ON-ORBIT INSPECTION, DETERMINING WHEN AN ON-ORBIT REPAIR OF THE TPS MUST BE ATTEMPTED, & DETERMINING READINESS TO COMMIT TO THE DEORBIT BURN AFTER A TPS ON-ORBIT REPAIR HAS BEEN ACCOMPLISHED. REPORT TO THE PRCB. DEFER - 10/29/04 SSP PRCB PRESENTER(S): JSC-EA4/G. GAFKA -----

S062375 BASELINE SHUTTLE SYSTEM INTEGRATION PLAN JSC-MS (SIP) FOR PRE-LAUNCH AND ASCENT DEBRIS CERTIFICATION WITHDRAWN PRESENTER(S): JSC-MS-----

S062383 EVA IR CAMERA JSC-MV JSC-MV/1-1 SUBMIT A SUPERSEDING CR TO ADDRESS FUNDING REQUIREMENTS AND STANDARD DEVELOPMENT SCHEDULE FOR THE EVA INFRARED CAMERA. REPORT TO THE PRCB. DEFER - 10/29/04 SSP PRCB PRESENTER(S): TBD -

**This is you!**

**Make it count!**



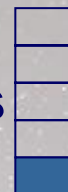
# Conclusion

- **Technical Wizard Success Mandatory Requirements**

- “Hard” technical skills



- “Soft” People Skills



- **Leadership Success Mandatory Requirements**

- “Hard” technical skills



- “Soft” People Skills



- **Success =**

- Loving what you do today (adding recognized value),
  - Knowing what you want to do tomorrow (adding recognized value),
  - Knowing how to get there,
  - Enjoying the journey along the way.

**I wish you your own personal situational success! Thank you!**